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4

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At left, main room in new basement laboratory showing tanks holding apparatus. Backward machine, furnace, and an inspection table for wires, special lighting. Top view is panel showing the office of the Civil War Museum, with photo microscope equipment and entrance to photographic dark room.

A portion (below) of the Chemical Laboratory which is equipped with the most modern apparatus for chemical control.

Close-up of operation of hydrolic testing device with test being made using an extensometer. A beam-type machine is in background.

AVIATION THE GLOBE AMERICAN RECREATIONAL MAGAZINE

THE CHINESE JOURNAL OF LINGUISTICS, 35(1), 1-12, 2002.

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RESEARCH DESIGN

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2/1/00	RENT	20.00	30.00
2/15/00	PAYROLL	50.00	20.00
3/1/00	RENT	20.00	0.00
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LOOK TO LOCKHEED FOR LEADERSHIP

AVIATION
August 1958
11



Picked Up Along Editorial Airways



ALABAMA MEETING: The day we dropped in for a visit with Barlow at his Alabama factory we found a couple of old flashes there, too. Jack O'Hara, down from Michigan, and Tom Mays, and Tom Mays (right). Both are now dispatchers for Barlow. Shortly after, we had a ride in the "Private Transport" with Jack Kelly, Barlow's Radio Manager.

IN A WESTERN COAST is a good idea for anyone, but an especially good idea for anyone in aviation these days, for our shore waters are currently becoming along in a big way. All factories are full of orders, both for domestic and export purposes, and all of them are closing the doors and looking up their operations to take care of the orders which will shortly be coming out of Washington for our Air Force re-building program.

IN AFTER SITTING IN at the Malabar Malabar Meeting (6/22 to 6/24), we spent the rest of the week going the rounds of plants in the Los Angeles and San Diego areas. We would like to take the opportunity here to thank all those at Longhams, Lockheed, North American, Consolidated, Ryan, and Sikorski for the courtesies extended to Charlie McKeown and your editor during this time. We hope

to see you all in the next before too long. How about coming over to see our Ford?

IN AND SPEAKING OF TADA, Tom Mays and his associates are to be congratulated for the job that they have done in presenting aviation to the public at the New York Fair. The task for putting the exhibition together was short and there were many difficulties in the way, but the committee succeeded in creating a degree of cooperation in the industry that was noteworthy. It is only to be regretted that since actual flying could not have been arranged for last visit, but possibly when North Beach airport is completed that situation will be greatly improved.

IN A TURNING POINT in air transport history is marked by the early summer of 1939. At long last after

AVIATION
August 1958
11



The T8P1 TRANSPORT [8-10 PLACE]

*Especially designed for
small field operations*

TRANSPORT • BUSINESS CRUISER • AIRAMBULANCE • FREIGHTER PLANE • EXPLORER • FREIGHTER
BOMBER • CHARTER PLANE • STAFF TRANSPORT • PHOTOGRAPHIC PLANE • INTERMEDIATE TRANSPORT

THE BARKLEY-GROW AIRCRAFT CORPORATION DETROIT, MICH., U.S.A.

years of development, of trial and of hard-earned experience, the Barkley-Grow is being introduced by regularly scheduled air service for passengers and cargo. In the March issue of *Airways* we set up the background against which Pan American is now sending its Clippers back and forth between Europe and America. There we wish only to mark the opening of that service, and to wish them God-speed as they embark on the final stages of another high adventure for American wings, the economical conquest of the last barrier, the North Atlantic.

AN OCEANIC ALIAS to that other organization which is now preparing to fly the same route, American Export Air Lines. Elsewhere in the issue will be found the news story of their first survey flights with the flying boat "Trans-Atlantic". With many years of seafaring experience behind it, American Export brings in the Atlantic air picture a type of experience that should serve them in good stead. To them, also, the best of luck.

ATTENTION, LADIES! If you are one of those unhappy souls who broods aside in despair and prolonged shames from home (necessarily on business), you are eligible to join Betty Lockew's new club—*La Ligue des Femmes des Pilotes en Aviation, Grouper Douglas*! (Which, being loosely translated, means "League for the Women of Men in Aviation, Four Thrusts"). Only requirement is that your husband must be away from home (ideally engaged in connected business) for at least one month of the year. Why not acquire a "Grouper" in your town? Why not get yourself elected an official "Keeper of the Nest Egg"? For full details, write Betty Lockew at 200 Riverside Drive, New York.

BIRTHDAYS seem to be cropping up thick and fast this month. The Army Air Corps celebrates its 26th Over 200, Donald Douglas and Walter North join the ranks of the real old timers with 25 years of arduous aviation, and TWA marks the end of its first decade of operations on its transcontinental route.

ANNIVERSARIES WERE TENDERED to Walter North at a dinner given July 11th at Walden's Alibi Hotel by Walden's Chamber of Commerce to mark the 25th anniversary of his first airplane flight. Many may claim the defection of first flights as long ago as that, but not many can match Deke's record for continuous

participation in and contribution to the aviation industry.

DOUGLAS DOUGLAS began his career during five years of service in the military in early war. Many tough years, and a long, hard pull by between the unfulfilled problems of MIT of the Class of 1914 and the war job of the largest aircraft manufacturing organization in the country. Douglas has left his mark on military and commercial aviation all over the world, from the time of the Battle-of-the-Worlds flight of 1924, down to the current Army bombers and DC series transports.

WIN FOR TWA becomes a hot hot month in Transcontinental Airways. After completion in first decade. Pioneered by a full-time group that believed it feasible to link East and West Coast by overnight or one-day service, the years have seen their

hopes become reality. In 1929, the predecessor company, TAT (Transcontinental Air Transport) began a transatlantic aerial service. That year it carried 14,000 passengers in Ford Trimotors (remember them?) over 2,000 miles of route, making the NY-LA trip in time 48 hours and 2 minutes. Today, as Douglas transports whisk its customers cross-country in 15 hours and 34 minutes. One year an estimated quarter-million people will travel over its 5,800 miles of line. Progress or something!

CONGRATULATIONS ALL! While on this festive mood *Airways* is tempted to whip up a cake and to add it with cordials—duty for the Army, twenty-five each for Donald and Walter, half a score for TWA,—and any number more for any of the rest of our good friends who happen to come aboard with a birthday.



"Napoleon's been out collecting royalties for his bubble gum of the Pacific Coast!"

Up... *smoothly!*



Because you start and end each flight with ground-contact, your landing gear has a doubled responsibility. And because this problem is our prime concern, Bendix gives it the concentrated, serious study it deserves. The result is a combination of highly efficient shock-absorption and smooth, easy landing and positive stopping.

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SHOCK STRUTS, TAIL SNOOGLER ASSEMBLIES

AVIATION
August, 1959

Side Slips

By
ROBERT OSBORN

At the moment of going to press the Far Eastern situation seems to include air fights over an extended length of border between Soviet-Mongol airplanes and Japanese-Manchurian airplanes. According to dispatches from Moscow about 100 enemy planes are brought down each day with the loss of only three of our own—with exactly opposite results coming from Tokyo.

As we should like to keep up on astronomical affairs throughout the world, even when they involve the unfortunate situation of an undeclared war, we wish some research organization would determine the proper percentages to use when reading these dispatches.

Now that Mr. Richard Airdhold has returned from his Spring boat exploration tour of the South Sea Islands we shall be very much interested in hearing if he discovered any new island which hadn't been trod. If perhaps he did find one we'd appreciate it very much if he'd whips at latitude to us soon, as we're



still looking for that beachcomber job for the duration of the next world war.

the expedition will be flown from Floyd Bennett Field to Flamingo Airport, they will be taken to the Aviation Building at the New York World's Fair. They will be received at the Fair by a welcoming delegation and will have an honor escort of Mounted Marshall Indians."

This information made us very sure that we hadn't been selected as a member of the expedition, at our one overwhelming ambition has always been to have as honor escort of Mounted Haskell Indians.

Now that the glider pilots at Ellerslie, N. Y. are making altitude records of over 18,000 feet, we think it is reasonable to assume that the engines which are not installed in their gliders are researched engines.

18 Across looking over the Boeing Clipper at Port Washington, just before the start of the regular transatlantic service. The intrepid Avianes remarked it was probably just as well he didn't have an airplane that big during his barnstorming days. Of course he would have been able to carry more passengers on each trip, but he noticed that the plane had a gross capacity of 4200 gallons and he doubted if he could have climbed that much out of one burner.

should like to predict a news story of a few more hours.

"The old flying boats, which were replaced on the ocean runs by much larger and faster equipment, are now doing good service on the 'Inland' in Central America, hauling mail, machinery and supplies to the island lakes and bringing out the commodities and a miscellaneous cargo including cypress logs. One or two of the flying boats are still being used by Pan American as training ships for the much larger equipment now in use."

DAWSON, GEORGIA—Wayne W. Friesen, Alcohol Tax Unit, is making it tough for the moonshine whiskey industry by spotting the stills from the air, then communicating by radio to other enforcement units on duty following along the roads. They find that system to be much more effective and have succeeded in catching a large number of stills operating full time.

Of course it's a poor rule that doesn't work both ways, so we suppose the airplane manufacturers will soon be selling similar equipment to



study customers of these mountainous hills. Magic can be possible to sell the mountainview of few permit shops around with approved rules.

While we haven't the slightest doubt that the intentions of the Bushinians are of the highest, we think that world conflicts are precarious enough without adding the risk of any more good-will fights. We dislike mentioning this complaint so often, but couldn't we recall that all this world unrest and Jewish roaming began when the boys started making their few good-will fights about 1330?

AVIATION
August 1978

1

30 Years

OUR AIR CORPS CELEBRATES A BIRTHDAY

Young!

Not quite thirty-two years ago, the Chief of the Signal Corps of the United States Army issued a specification for a flying machine. Unlike most such specifications today, it was probably written without the least expectation that anyone could possibly meet it. It called for an aircraft of unbounded performance. It was to carry two persons, with sufficient fuel for a flight of 125 miles. It was to remain in the air for at least an hour, and during that time it had to be steerable in all directions without difficulty, always under perfect control, after which it was to return to the starting point and land without any damage. A speed of 40 m.p.h. was desirable, with nothing less than 30 m.p.h. acceptable. Furthermore, it had to be quickly

and easily taken apart for transportation in Army wagons (how the old Army mules must have laughed!) and put together again in operating condition in one hour.

Such rigid specifications may well have stumped the constructors of the day, all but two,—a pair of brothers hailing from Dayton, Ohio. By 1908 the Wrights set up at Fort Meyer, Va. a machine which might easily have met the requirements but for a minor fault. First result was tragedy,—the death of Lieut. Tom Selfridge, serious injury to Orville, and complete loss of the machine.

Not discouraged, by late spring 1909 they had another machine in readiness, and on July 30, with Orville

Wright and Lieut. Benjamin F. Cook up, the airplane successfully passed its speed trials. Later in the day, with Orville at the controls and Lieut. (now Colonel) Lahm as passenger, the machine set a world endurance record of one hour, twelve minutes!

Success! The specifications were met and the Army gave new possession of its first airplane. To complete the contract, two officers were taught to fly—Lieut. Frederick E. Humphreys and Lieut. Frank P. Lahm.

Thus was born the United States Army Air Corps. This was the modest beginning of the first thirty years of service in the country that our Army flyers have just completed, a history that has included active participation in two wars,—a little one in Mexico, a big

one in Europe. No less honorable has been its record in times of peace.

But it is a far cry from the flimsy Wright machine of 1909 to the Flying Fortress of today. An American we can view with justifiable pride the technical developments, and the Service testing of three decades which have stepped speeds from a drabbed forty to a dizzy 400 m.p.h., ranges from a bare hundred to thousands of miles, and payloads from 300 pounds to many tons. We pride also we may look upon a powerful that ranks with the finest in the world.

Congratulations, Army! You have on your eye teeth. Your adolescence is behind you and you now face the future in the full strength of your maturity.

The WRIGHT BROTHERS as Researchers

An appreciation of their early contributions to the aeronautical sciences.

Being an excerpt from the 27th Wilbur Wright
Memorial Lecture read before The Royal Aero-
nautical Society, London, May 22, 1939

By George W. Lewis, F. I. Ae. Sc.

Director of Aeronautical Research, R.A.F.

THESE personal relationships with Orville Wright and through investigations of the early history of aviation it has been my privilege to through some estimate of the scientific qualities and contributions of the Wright brothers. In their day the accomplishment not only of flying the first successful power-driven aircraft, but of constructing the research equipment to promote design data and of constructing both airplane and engine themselves was indeed remarkable. The methods and results deserve the term "modern" in the highest sense, in that they were far in advance of the contemporary art, and served as the basis for unprecedented practical applications.

Through the courtesy of Orville Wright, some information on the Wright brothers' wind tunnel and the

researchers conducted is available to me, and this subject will comprise the first section of this paper.

Fig. 1 shows two views of a replica of the original Wright brothers' wind tunnel. The propeller was mounted in the center of the circular wing past apparatus of the straightening vanes. The tunnel itself was 5 feet long, and the test section was square in elevation, being 22 inches on a side. The speed attained was about 27 miles per hour, or 40 feet per second.

Fig. 2 is a reproduction of 1 quote Orville Wright. The original photographs

(made December, 1907) giving the profile and plan form of most of the airfoils mounted in the wind tunnel in 1907. In this figure 1 wish to write your attention particularly to models 7, 8, 9, and models 10, 11 and 12. Here it is shown the test program for the first investigation conducted in the United States on a systematic series of airfoils. Models 7, 8 and 9, as will be noted, are camber air foil plates with systematically varied camber, models 10, 11 and 12 have the same amount of camber in a more forward location. These models

had approximately 1 inch chord and 6-inch tips, giving an aspect ratio of 6, now standard for aircraft tests in most aerodynamic laboratories. Systematic investigations on the effects of plan form, aspect ratio, higher camber ratio and some miscellaneous studies were also conducted.

This is related a surprising research program for the year 1907. I wish to emphasize particularly the broad fundamental point of view it shows. So clear was the Wright brothers' appreciation of the basic factors in the problem that they included all but one of the important variables that have since centered in so wing and aerobal research. In this respect some subsequent research along this line appears mainly to have been an extension and refinement of the basic outline. Short it was never published, the research program clearly bears the independent approval of eminent research scientists in this field, who have almost universally employed the same method of attack.

We turn next to the balance system for measuring the forces. Fig. 3 shows a diagram and reproduction of a photograph of the original lift measuring balance. The legend at bottom, is Orville Wright's own description—

Lift Measuring Balance

Letters corresponding with parts shown in photograph and diagrammatic sketch, Fig. 1.

F—Frame. Mounted on floor of tunnel at 90 degrees to air stream.
D—Disk graduated in degrees.
P—Pivots mounted on side A.
A, A—Vertical scales.
B, B', B"—Horizontal arms mounted on scales A, A.
C—Cross bar swinging on arms B, B', B'.
R, R, R, R—Resistance surfaces and instead of a square plane to avoid deflection direction of air current striking surface S. They had no deflection arms themselves but together with the cross piece C had a permanent signal to that of a square plane of 8 square inch area, mounted at 90 degrees in the plane of the surface S.
H, H, H, H—Horizontal arms mounted on friction discs T, T.
T, T—Friction discs, sliding over axle A, A.
J—Cross bar supporting the surface S.
K—Balance is to be measured mounted on K at various angles, to be adjusted.
M—Mounting piece attached to J.

The pressure on the surface S tends to move the pressure P inward 40 degrees, while the pressure on K, R, R, tends to move the pointer back toward zero. Before making a reading the arms R, R, must be adjusted always through the friction sleeve T, T, parallel to the air stream. Then the size of the angle indicated by P, multiplied by eight and divided by the area of the surface being measured,

will equal the lift of the surface S in per cent of pressure on a normal plane of equal area.

The method of measuring drift or drag, was equally rigorous. An auxiliary balance, shown in Fig. 4, was used. This balance measured the value of drift to lift directly as the tangent of an observed angle—a method which may appear involving and slow to the wind tunnel investigator, nevertheless as he is with the collection of related coefficient effects. The photograph shows a replica of the original balance and the legend, as previously, is Orville Wright's description—

Drift (Drag) Measuring Machine

Fig. 2 reproduces diagrammatically, the D/L measuring machine.
H, H—Parallel arms mounted on scales A, A.
C—Connecting links between arms H, H.
S—Arm to be measured, mounted on connecting links C.

E—Disk graduated in degrees.
P—Pointer attached to arm H.
R—Resistance surfaces of all pins arms on arm H.
D—Resistance Drag.
Then an angle H, A, H, as indicated by pointer P, gives D/L.
F—Frame (see photograph).
In practice we modify related the (Turn to page 21)

Fig. 1. Replica of the Wright Brothers' wind tunnel.

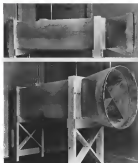


Fig. 3. Reproduction from the original research program (1907).



Fig. 3. Sketch and photograph of the lift measuring balance (1907).

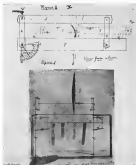


Fig. 4. Sketch and replica of the D/L measuring balance.

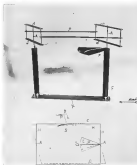




Photo by the author and another pilot



2



3



5



6

Soaring's

The 10th Annual National Soaring Contest.

A successful meet with a flock of new records in the bag has created new enthusiasm and paved the way for a new era of soaring development in America.

By Maurice A. Garbell

Member, Executive of the A.S.C. Committee for Maritime Flight

THE 1938 soaring season has given to the U.S. its first three Golden C's. To win these gliders must make a distance flight of at least 366.4 miles and reach an altitude of at least 15,225 feet above point of release. There are only about 25 Golden C's in the whole world. The impulse of the International Research Committee for Maritime Flight in establishing this award, was to develop ability in cloud flying among soaring pilots. Under normal European meteorological conditions, and also under the conditions normally found along the American East Coast, it is almost impossible to gain more than 5,000 feet without drifting through a cloud.

It is interesting that after so many years when the American altitude record seemed to work best as a figure around 6,000 feet, three gliders succeeded in breaking through it to set a new mark three times higher than the old one. It is also remarkable that three very different gliders were used for these flights. Only Doctor's *Maquise* belongs to a type of sailplane previously scored in altitude flights. Eugene Robert Stouley's light structure at Pensacola, reached the new American record mark of 12,625 feet on his new ship *Maquise*, while John Robinson of San

1. One of the best and efficient American sailplanes.
2. Eugene Robert Stouley's light structure at Pensacola.
3. Earl Stouley's *Maquise* on its first flight.
4. The A.S.C. sailplane that achieved altitude in the 1938 meet.
5. A lot of heavy backing on a sailplane's fuselage.
6. Earl Stouley and John Robinson manage to keep their glider on the water.

Decennial

Elmira, N. Y., June 24—July 3, 1938.

Drum, Cal., gained 15,000 feet altitude aboard his *Zeus*, a multi-faceted job based on a *Grasshopper* wing. Chet Doctor's altitude was about 10,000 feet. A new American altitude record for two-seaters was also set during the contest by Burringer, former General Manager of the Soaring Society of America, on the *Schweitzer* off-metal sailplane. On July 4, he reached 6,710 feet above point of release with a passenger aboard.

All these altitude flights have been made by utilizing powerful cumulus clouds. They have been made possible by heavy cloud flying training of the pilots and by better ideal flying arrangements on the days.

The highest altitude ever obtained on a sailplane plane (10,000 ft.) was reached by the help of cloud-groomed currents on which it is possible to reach high altitudes without entering turbulent clouds. Wave patterns are found behind mountains and help in certain weather conditions, but the flights in Elmer had to be done exclusively under, and within cumulus clouds. This fact gives even more value to the altitude flights performed this year.

As for distance flights, a particularly difficult record was developed around the goal—200 miles. The 100 miles from Elmira, which was finally won by Bob Stanley, on the *USA No. 1* in 4 hrs. 36 min. The greatest distance ever flown in an Elmer contest was Chet Doctor's flight in his *Maquise* on July 2, when he landed at Atlantic City, N. J., 232 miles away. This figure, however, is not up to the American record of 280 miles, recently established by Woody Brown during the Wichita Falls contest in Texas.

Distance flights nowadays have lost practically all their technical interest. It is considerable, however, that the soaring pilots at Elmer normally have about seven hours

7. Chet Doctor in the machine in which he made the Atlantic City flight.
8. Harry Stevens lands over his *Elmer* glider.
9. The *USA No. 1* in its first flight.
10. The *USA No. 1* in its first flight.
11. The *USA No. 1* in its first flight.



sonel, A. Trumbull, foreman, radio & electric shop, C. M. Peterson, foreman, propeller shop, E. A. Bennett, foreman, wheel overhaul, E. Wray, foreman, machine shop, D. Stoughton, foreman, sheet metal shop, R. K. Moore, assistant foreman, J. Koller, foreman, battery shop, J. E. Leonard, inspector, C. A. Leonard, foreman, T. Williams, sea pilot, E. E. Parkash, assistant diesel engineer, H. E. Salisbury, assistant superintendent, repair line; W. C. Conard, foreman, engine overhaul; E. Thompson, helper maintenance; L. J. Korte, foreman, accessory department, A. J. Carrell, foreman, plane overhaul, L. D. Ross, foreman, instrument department.

"It Takes All Kinds—"

Many and varied are the activities at United States Marine Corps Depot No. 1. Most of these activities may be carried out on the shop yard on the preceding two pages. As for inside:

1. After arrival, each of the Pratt & Whitney 14 cylinder engines is given an "X-ray" test run, employing a calibrated test "fish." One of the two suspended, disassembled test stands. A check meter reads engine load.

2. These auxiliary tanks, developed jointly by United and Standard Oil, carry 1000 gallons of gasoline, 500 gallons of oil. They are fitted with two valves which ensure complete delivery under less than 100 pounds per square inch pump. United now has eleven such tanks tested at major service points.

3. An inspection department has already been established at Chryseus to handle new patterns and design work for the Ordnance and Supply stores.

4. Airplane maintenance, subsidiary repair, engine overhaul. Radi inspection, etc., are carried on in the Service Repair Shop. Chief of this section, Kenneth J. Barker, is interested not only in repair work, Assembly line to test, then use of accessory repair shops.

5. Plans require for instrument maintenance in a class shop. At Chryseus, the air training system provides standard air. The shop is divided into three sections: (a) overhaul and repair department for instruments requiring extensive precision work (b) machine assembly in general part of the test equipment; and (c) a third room contains the necessary machinery and storage for gauges, oils and other supplies.

6. Chryseus is the main distribution and control point for all arms and supplies sent into the radio system. A portion of one of the stock rooms.

7. The Hydrostatic pressure used on all United Marine Corps items through this shop for periodic check and overhaul. Many special tools and fixtures have been developed at Chryseus to simplify gun, pipe, aerobics and submersibles.



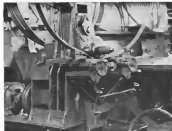
Vultee's Special Machines

In a previous article the author discussed attachments and fixtures for standard machine tool equipment. The concluding installment covers machines designed specially for aircraft work.

By Don I. Carroll

Machine processing operations in the Vultee plant are rather sharply divided between the machine shop proper and its other applications of machinery to airplane building. The latter are frequently novel or highly specialized, standard machine tool operations were discussed in a previous

issue. The group of air machines covered in this story is that which is used in the machine shop proper and in the aircraft building department.



Below is a machine designed from standard stock of lathe-turned sections. Designed and built in the Vultee plant, this machine rolls elements from flat stock and rounds or chamfers deeps from turned or milled stock.

article. The numerous special aircraft machine equipment can be subdivided into three general classes. The first is profiling equipment, which is used for cutting flat stock or extrusions accurately in a predetermined pattern. The second, turning equipment, such as presses, drop hammers and power hammers, which are used for shaping sheet metal into various, and often quite complex, forms. Finally, there is the assembly equipment, which includes special production jigs, special

forming machines and electric-welding machines for seam and spot welding.

In our factory shops we have developed the power hammers, hydraulic press, punch press, drop hammer, and other such equipment for the economical manufacture of complex sheet-metal parts. The power hammers are used for forming angles and various types of straight sections from flat sheet material. The hydraulic press has found wide application in the aviation industry and is available for forming very large sheet parts, as well as many different small parts in a single step.

We use the hydraulic press for forming. Engines, and bearings in our operations, as an example. The hydraulic press is a mechanical variation of the hydraulic press. This tool is now being used by North American Aviation and may find application elsewhere. Its basic advantage is high-speed operation for certain types of pressing. The punch press, used largely for blanking operations on parts not greater than 12 in. across, is gaining popularity in connection with the Constantan Co. process. The Constantan dies are produced with the De-Al machine, making flat dies from cold-rolled boiler plate at 3 in. or 4 in. thickness. By this process it is possible to produce a flat and punch quickly with little hand work. The punch is cut and the setting is extremely simple. Cost of this method is about one-third that of normal die cuts and the work is entirely satisfactory for highest production runs. Forming and blanking operations are largely performed with the punch press, although many simple forming operations are also done



The hammer assembly bay with its roller top.

The drop hammer is in some respects our most versatile forming tool, making possible the production of some parts which could hardly be manufactured by any other process. The presses and drop hammers together make it possible to use double-curvature sheet-metal parts that would otherwise be too expensive; thus, they contribute to the aerodynamic efficiency of the airplane as well as to economy of production.

Machine men generally are familiar with the new aircraft drop-hammer techniques, which employ die and lead dies in connection with a flat-bed hammer to stamp shapes in sheet-metal or mild-steel material. We have introduced our share of improvements to the technique, including use of multiple progressive dies for deep forming operations, roller blades or pins for forming round and oval shapes, and many other detailed refinements of the operation. But we believe our greatest contribution is in our application of production methods to the entire drop-hammer process.

Double-Stroke Hammer

Also in the drop-hammer shop proper we have increased production per man hour through design and manufacture of our own hammers. They are distinguished by unusually rapid construction, with very heavy cast-iron beds, or anodes, and overhead "hammers." As is customary,

we use a rope lift working from an electrically operated electric crane, but for the sake of safety we use a double-rope lift system, and a double system of safety latches. This makes it physically impossible for the hammer to fall unless deliberately dropped at the will of the operator.

Operation of the hammers is under complete automatic control through the design of the support system for both power drive and hammer head. The former is supported on four heavy casters which swing and mounted in four driven rollers mounted from the base of the hammer. The hammer head extends out to great support to heavy vertical rods, rigidly attached to the

base, which guide the hammer head in its fall. This construction assures perfect alignment. The hammer base is mounted on a system of rollers which extend down to a substation concrete base. A system of eye bolts is attached to the hammer power-drive support pedestal so as to provide the operator with light means to reach into all parts of the die at work during stamping operations. Handwork work is further quickened through provision of work layout tables adjacent to the hammers in such a way that the work is accessible with minimum effort both before and after stamping.

One of the major bearing problems at Vultee has been the production of double or oval rings for bearing housings and other purposes. They could not find any economical satisfactory for this purpose and finally designed our own machine, which is relatively simple in construction, and extremely efficient in production. The machine consists essentially of a set of two rollers mounted one above the other and driven in variable speed in either direction. Alongside the lower roller are two tilting tables, accurately constructed, to feed the work to a desired curvature. The roller dies may be changed on their rotating shafts to provide for a wide variety of channel-shape forming.

With this machine we roll straight channel sections from flat strip sheet

(Continued on page 37)



Flanges for the center section of the wing are extruded sections rolled in the double profile. The machine is designed from standard stock machine tool equipment.

SIMPLIFYING Stress Analysis

By F. R. Shanley

Chief Engineer, Curtiss Wright Technical Institute
Structures Staff Engineer, Lockheed Aircraft Co.

IN view of the ever increasing dependence on the stress analysis of modern metal structures it is always good news to hear about any simplifications that can be effected. It may be worthwhile, therefore, to point out a few recent developments that may not have become generally known to stress-analysis engineers.

Load Axis.—In determining the structural loads and bending moments acting on a wing structure it is a good idea to become confused because of the presence of dihedral, sweepback, variations in elastic axis position, and other such factors. The use of an arbitrary "load axis" has special advantages in that it merely ignores the actual load distribution from the geometry of the structural structure. The load axis is simply a straight line drawn normal to the plane of symmetry of the airplane. It may be located at any convenient point, preferably somewhat ahead of the wing and either completely below or above it. Once located, it serves as a reference axis for all subsequent calculations of wing and tail loads. Standard conventions for loads and torques should be adopted, as indicated in Fig. 1.

Regardless of the methods used for load distribution and summation, it is possible to transfer all loads and moments to the load axis and to resolve them into components along the chosen reference axis. (Torques should of course be designated by vectors representing the axis of rotation.) This permits the computation of curves of shear, bending moment, and torque with reference to the load axis. In the actual process, the take load will include the distance from the aerodynamic center to the load axis at the various spanwise stations.

The main advantage of this system

is the fact that a great deal of computer work can be avoided before the actual stresses of the wing have been developed, and, in cross section, the elastic axis (as called) of the structure is not a straight line, the load-axis method eliminates the torque corrections that must be made while "moving" shear loads over a wavy line.

Another feature of the load axis is the fact that balancing tail loads are influenced only by balanced moments about this axis. It is therefore possible to utilize the total suspended torque about this axis for balancing purposes. In the procedure used at Lockheed the entire process is carried out in terms of assembly elastic axes, resulting in a set of ground curves for tail load in each element form. These curves require enlargement as long as the geometry of the airplane is not altered and they can be used for any design condition. Incidentally, the need for a mean aerodynamic chord disappears entirely. Since a structural standpoint, and it becomes possible to use an approximate M.A.C. simply as a check-point to measure c.g. position.

Dihedral and Sweepback Effects

After the wing loads have been determined with reference to the load axis they must eventually be applied to the basic wing structure. This usually involves a determination of some sort of shear center or structural axis (as proven that will be discussed in greater detail later in this paper). To transfer the actual loads to this shear center it is only necessary to make a correction in the torsional moment about the load axis. The loads and torques now act in the right place, but not necessarily in the right direction. What are the

DI model shown in structural equations, as it represents results of actual application of modern stress analysis methods in preferred position. The entire set of equations of the Curtiss-Wright Technical Institute at Glendale, Calif., and Princeton Staff Engineer for Lockheed Aircraft, Burbank. We are pleased to call this article to the first list of available technical contributions which the Society has made to AVIATION.

right direction? This simple question is not so easy to answer in actual cases, but a few elementary principles can be applied to clarify the point.

Let us assume a condition such as shown in Fig. 2, where the solid lines represent the basic wing structure. The vectors α , β , and γ represent either loads or torques. The first and most obvious principle to apply is that the internal loads should be found by "cutting" the structure through the maximum cross-section. In Fig. 2 the proper cutting plane is illustrated by the lines A-B and C-D, which can be regarded as traces of the plane. The determination of the cutting plane in this manner simply amounts to assuming that the structure behaves in the same way regardless of how it happens to be attached to the fuselage. Obviously it will be necessary to resolve forces and torques into different planes and directions in the fuselage attachment, and there will also be some question of "end effect." But for most studies outboard of the fuselage the structure should be regarded as independent of conditions in the rest of the wing. (In some cases the maximum cross-section principle may have to be further modified, such as when

struts do not run normal to the cross-section.)

The final step is, of course, to resolve the loads and torques into new planes determined by the plane of maximum cross-section. This can be done at the same time the external loads are moved to the structural axis. But when the dihedral or sweepback angle is not large it becomes advisable to determine the stress that would be introduced by ignoring these effects entirely. It will be found that these errors can be reduced by the simple correction factors in most cases.

Corrections for Dihedral and Sweepback

Knowing the correct procedure, it is possible to evaluate the errors introduced by carrying all actions parallel with the plane of symmetry. The approximate correction factors have been tabulated in Table 1. It is arrived at these values certain assumptions were made, such as:

- (a) Wing area used in load factor calculations is the projected area.
- (b) The air pressure acting over a curved area of wing is unaffected by dihedral. (This assumption results in the correct total vertical force.)
- (c) In computing section properties the skewed areas of struts, corrugated lips, etc., are used, even though the section is taken at an angle with the struts.

It can be seen from Table 1 that where the angles involved are small errors in the correction factors become negligible. This enables the analyst to be made with a minimum of work. It will often be sufficient to make a correction for only one or two cases.



TABLE 1

Correction Factors

		Actual Dihedral	Sweepback
Wing loads			
Vertical force	P	$V \cos \alpha$	$V \cos \beta$
Horizontal force	H	$H \cos \alpha$	$H \cos \beta$
Bending moment	M	$M \cos \alpha$	$M \cos \beta$
Shear	S	$S \cos \alpha$	$S \cos \beta$
Twisting moment	T	$T \cos \alpha$	$T \cos \beta$
Strut loads			
Vertical force	P	$V \cos \alpha$	$V \cos \beta$
Horizontal force	H	$H \cos \alpha$	$H \cos \beta$
Bending moment	M	$M \cos \alpha$	$M \cos \beta$
Shear	S	$S \cos \alpha$	$S \cos \beta$
Twisting moment	T	$T \cos \alpha$	$T \cos \beta$
Section properties			
Area	A	$A \cos \alpha$	$A \cos \beta$
Second moment of area	I	$I \cos \alpha$	$I \cos \beta$
Product of inertia	J	$J \cos \alpha$	$J \cos \beta$
Area moment of inertia	I_x	$I_x \cos \alpha$	$I_x \cos \beta$
Area moment of inertia	I_y	$I_y \cos \alpha$	$I_y \cos \beta$
Area moment of inertia	I_z	$I_z \cos \alpha$	$I_z \cos \beta$
Area moment of inertia	I_w	$I_w \cos \alpha$	$I_w \cos \beta$
Area moment of inertia	I_v	$I_v \cos \alpha$	$I_v \cos \beta$
Area moment of inertia	I_u	$I_u \cos \alpha$	$I_u \cos \beta$
Area moment of inertia	I_t	$I_t \cos \alpha$	$I_t \cos \beta$
Area moment of inertia	I_s	$I_s \cos \alpha$	$I_s \cos \beta$
Area moment of inertia	I_r	$I_r \cos \alpha$	$I_r \cos \beta$
Area moment of inertia	I_q	$I_q \cos \alpha$	$I_q \cos \beta$
Area moment of inertia	I_p	$I_p \cos \alpha$	$I_p \cos \beta$
Area moment of inertia	I_o	$I_o \cos \alpha$	$I_o \cos \beta$
Area moment of inertia	I_n	$I_n \cos \alpha$	$I_n \cos \beta$
Area moment of inertia	I_m	$I_m \cos \alpha$	$I_m \cos \beta$
Area moment of inertia	I_l	$I_l \cos \alpha$	$I_l \cos \beta$
Area moment of inertia	I_k	$I_k \cos \alpha$	$I_k \cos \beta$
Area moment of inertia	I_j	$I_j \cos \alpha$	$I_j \cos \beta$
Area moment of inertia	I_i	$I_i \cos \alpha$	$I_i \cos \beta$
Area moment of inertia	I_h	$I_h \cos \alpha$	$I_h \cos \beta$
Area moment of inertia	I_g	$I_g \cos \alpha$	$I_g \cos \beta$
Area moment of inertia	I_f	$I_f \cos \alpha$	$I_f \cos \beta$
Area moment of inertia	I_e	$I_e \cos \alpha$	$I_e \cos \beta$
Area moment of inertia	I_d	$I_d \cos \alpha$	$I_d \cos \beta$
Area moment of inertia	I_c	$I_c \cos \alpha$	$I_c \cos \beta$
Area moment of inertia	I_b	$I_b \cos \alpha$	$I_b \cos \beta$
Area moment of inertia	I_a	$I_a \cos \alpha$	$I_a \cos \beta$

Chord Coefficients

A considerable simplification in the determination of "dihedral" loads can be effected if we make use of a simple aerodynamic chart. If we know the lift coefficient for a given section or strip of wing, the corresponding chord coefficient is immediately determined and is unaffected by aspect ratio. This follows from the fundamental assumptions made in determining

(Turn to page 47)



DE HAVILLAND FLAMINGO



Success as a general purpose transport for the De Havilland Flamingo is available in countries of 12, 17, and 20 passengers for operation at wheels, floats, or skis. It is a low-wing monoplane. Equipment includes Handley Page slotted flaps which increase lift 80% about 40 per cent when fully lowered and may be used for take-off at a setting of about 20 deg. at which point the lift increases to about 25 per cent.

Simplified maintenance is gained by extensive interchangeability of structural parts such as the two landing gear assemblies, port and starboard engine installations, and others. One of the two 600 hp. Bristol Perseus engines which engines may be replaced within an hour.

Below: Values in overhead and again provided in the photo cockpit in the nose of the Flamingo.

Left: There is ample room for the comfortable maintenance and sufficient room space in the main cabin.

The soundproofed cabin is exceptionally strong being 7 ft wide and 6 ft, 7 in. in its maximum height. Wide crossbars are placed close abreast with ample aisle space between rows. The entrance door is only 27 in. above the ground. The 12 passenger version carries a crew of three and has a block-to-block range of 1300 miles. For twenty passengers, a crew of two is specified and the block-to-block range is 600 mi.

General specifications are as follows:

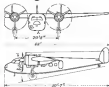
Span, 48 ft. (14.6 m.)
Length overall, 35 ft. 7 in. (10.9 m.)
Height overall, 13 ft. 4 in. (4.0 m.)
Wing Area (one engine version), 320 sq. ft. (29.7 sq. m.)
Gross weight, 12,000 lb. (5443 kg.)
Payload (1200 mi. version), 2111 lb. (957 kg.)
Perseus (1200 mi. version), 2300 hp. (1690 kw.)
Maximum speed (1200 mi.), 262 m.p.h. (423 km/hr.)
Cruising speed (1000 mi.), 237 m.p.h. (381 km/hr.)
Service ceiling, 16,500 ft. (5000 m.)



The landing gear wheel with air deflationable and the three blade De Havilland propellers provide ample ground clearance.



Not only with the plenty of extra bench space airplane leg room even when the main cabin is in military position.



New Goodrich E.T.* Brakes Landings To All Planes



CLOSE-UP VIEW OF THE 20-INCH GOODRICH E.T.* BRAKES ON A BOMBING "FLYING FORTRESS"

GIANT BOEING B-17 TYPE "FLYING FORTRESS" EQUIPPED WITH 20-INCH GOODRICH E.T.* BRAKES . . .

One of the most demanding planes built by Boeing Aircraft Company for the United States Army Air Corps, the Goodrich E.T.* Brakes used on these giant bombers have 25 times as much braking surface as the conventional air brake of America's previous light sport plane.

HOW GOODRICH E.T. AIRPLANE BRAKES WORK



1 Cross-sectional view showing the Goodrich E.T.* Airplane Brake installed. The Expander Tube (A) is deflated, the Exposed Brake Block (B) retracted, and the wheel (C) free to turn.



2 The Expander Tube (A) is inflated with landing fuel (A), forcing the Exposed Brake Block (B) against the brake drum (C), bringing the plane to a quick, smooth, safe stop.



3 The full circle of sectional brake blocks (B) is joined together by metal expansion springs (D). When the Expander Tube (A) is inflated, these brake blocks press evenly against the brake drum (C) all the way around. When the Expander Tube deflates, the springs retract, pulling the brake blocks back from the drum and leaving the wheel free to rotate again.

The New Goodrich

Over 40 Rubber Products for Airplanes—including Tires—Tail Wheels—
—Rubber Hose—Greenacres—Shock Absorber Cord—A Complete Line

AVIATION
August, 1939
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Give Smoother, Safer Whether Large Or Small



1938 PIPER CUB COUPE EQUIPPED WITH 5-INCH GOODRICH E.T.* BRAKES

A lightweight, safe, easy-to-fly plane that provides extra plus safety margin in an emergency. Among other assets, this aircraft has the safety and convenience of landing with Goodrich Airplane Brake technology and the new Goodrich E.T.* Brakes.

New Expander Tube Principle Provides Extra-Safe, Extra-Smooth Landing Control for Every Size and Type of Plane

TODAY, everything from half an ounce plane to "Flying Fortresses" are landing smooth, safe, smooth landings on the proven combination of Goodrich E.T.* Brakes and Goodrich Airplane Brake technology. And it's the new simplicity of the new brake design that does it.

The new Goodrich E.T.* Brake has only three basic elements: the Expander Tube, the brake block, and the return spring. Working together, these simple but good elements are absolutely dependable and economical compared to other brake systems.

Today, everything from half an ounce plane to "Flying Fortresses" are landing smooth, safe, smooth landings on the proven combination of Goodrich E.T.* Brakes and Goodrich Airplane Brake technology. And it's the new simplicity of the new brake design that does it.

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CUB OWNERS GET EXTRA-SAFE LANDING CONTROL WITH THE 5-INCH GOODRICH E.T.* BRAKE SHOWN ABOVE . . .



E.T. Brakes

Also See Shoes—DEICERS—Mating
of Rubber Aeronautical Accessories

AVIATION
August, 1939
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Bowlus Baby Albatross

In his years of motorless aircraft design experience Bowlus has evolved many types, but none so drastically American, and so delicately elegant, as the Baby Albatross. Bowlus is not building complete planes, but prefers to stress a kit plan which provides the purchaser with parts that are 50% completed in the factory, but on which all the detailed finishing, assembly, etc., is left to the individual builder, or club. Under this plan the kit is supplied in ten units; the builders starting with the simplest part of the ship, the molder, and progressing through the elevator struts, fins, etc., until the plane and its builders have progressed together to the point where one is a finished article, the other a finished workman. This procedure lowers the cost of the plane materially, teaches the purchaser many things that he should know about his plane before he tries to fly it,

and has a better purchase for continued maintenance of the plane, and greatly widens the possible field of usefulness connection through providing interested groups with a properly engineered design, and with parts that are 50% factory finished.

Under the Bowlus purchase plan the complete kit may be bought as two increments for a total cost of \$285. For another \$5 one can obtain complete instructions of the building trailer that has been developed for transportation on the ground. The complete plane factory finished, may be had for \$500. The approved purchaser also will get a \$15 down payment, which covers two units, the molder, and elevator. A second payment of \$30 covers units 3 and 4, the wing struts and the fin unit. Payment of \$145 completes the larger item, the two wing units, Nos 5 and 7. These come with the spar and landing wings complete

except for finishing, and with all ribs assembled, ready for attachment. In fact, throughout the kit, all major structural units, such as pod, tail boom, and wing spars, are finished at the factory to assure proper strength of the plane's primary structure. Unit 5, the ailerons, costs \$25. And units 8, 9, and 10 are lumped in a \$120 payment, covering the pod, boom, and miscellaneous finishing parts and materials. All fabric coverings are sent at the factory ready for installation. Detailed instructions are supplied throughout.

In general design the Bowlus Baby Albatross is a straight-tail high wing monoplane of wood construction. There is no fuselage prop, pilot accommodations being provided by an abbreviated nacelle, or "pod", all three-piece wooden chord and ground in a mold to a perfect streamline form and glass-like finish. The pod has a landing

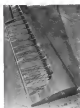
skid and take-off wheel. Tail surfaces are carried on a fixed boom attached to the pod through three laminated doublers (lips). Wing is of monospar type with a single strut for each panel.

The wing is of elliptical plan form, and is tapered in thickness from the forward end of the ailerons, although straight section to that point. Aired section is Gossamer 150 half way to the tip and then modified to a symmetrical section at the tip. No spacers or ribs are used as it is claimed that the Baby Albatross will land at 27 mph, and will do a three hundred and sixty degree turn in ten seconds.

Finings throughout the Baby Albatross are of mild aluminum alloy. Span of the ship is 44 feet and weighs 265 lb. Climbing rate is one mile 20 to 1, and best flying speed is 40 mph.



The construction of the Baby Albatross



The laminated wing doublers

Briegleb Utility

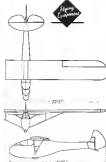
One of the most promising and most designs from the standpoint of commercial application, is the latest Briegleb Utility. Glider of rugged design for student instruction work, but with strong soaring potentialities. Of very light empty weight, and streamlined small dimensions, the Briegleb B-16 is a conventional strut braced general monoplane type, with fully enclosed cockpit. Differentiated ailerons are sufficiently sensitive to provide lateral control at only 30 mph. A push-pull system is used for aileron and elevator controls. The plane may be set up ready for flight by two men in twelve minutes. 1000 steel tubing is used in the fuselage structure. A 12 in commercial wheel and tire are used for landing and take-off, in addition to a nose skid and a tail skid. The glider is designed for a top speed of 50 mph for airplane towing.

A sand-slice glass has been worked

out to encourage home builders, or club groups who wish to assemble the ship themselves. A variable purchase plan includes the following parts list: Complete plan, working drawings, and detailed instructions, \$12.50. Complete plan and all new materials, \$175.00. Plan, materials and all metal work finished, including all welding and fitting, fastenings, etc., \$175. Complete set of finished parts (the main set, all ready to assemble \$140.00, or the complete Utility, Brierley-built, \$485. A further step has been developed by which groups may purchase the parts in five or ten units. Under this plan the plane is subdivided into such component parts as ribs, ailerons, elevators and molder, etc., and these units may be supplied step by step until the job is ready for assembly, spreading purchase over a period of five or ten months on the basis of one unit per month, or even longer if necessary or desirable.

Briegleb B-16 specifications

Span	22 ft. 3 in.
Length	36 ft. 10 in.
Area	NACA's 4412
Aspect ratio (estimated)	6.5 to 1
Wing area	117 sq. ft.
Empty weight	250 lb.
Loaded weight	320 lb.
Wing loading	12 lbs. per sq. ft.
Glider ratio	16.5 to 1
Glider speed	22 ft. per sec.
Stalling speed	30 mph



Finished fuselage and the detailed pod of the Bowlus Baby Albatross





New Beechcraft

Model 185 has 11,950-hr. ceiling with Full Load and One Dead Engine

SINCE engine performance of twin engine ships has been qualified in so many ways that when a manufacturer comes out with a ship that will fly anywhere in the country with full gross weight and one engine dead, we cannot help believing he has something. Such is the case for the new Model 185 Beechcraft. Its C. A. A. ceiling rating is 12,300 ft. with full gross of 7500 lb. Power plant equipment includes 450 hp. Pratt & Whitney Wasp Jr. engines, Hamilton Standard constant speed propellers, Cessna edge fuel analyzer and 20-gal. fuel capacity.

The model 185 Beechcraft introduces a new nose and NACA cooling combination and a new tail for Beechcraft twin motors. By means of these improvements it has been possible to secure very satisfactory sailing characteristics. At about seven or eight miles per hour before the airplane stalls, the nose begins to hop up and down a little but the control column moves backward and forward slowly through a very small arc while giving the pilot an automatic warning of an approaching stall. Does it be three yards the warning and continues to slow up the airplane until it stalls. It does not drop off on red wing and merely nose down fairly slowly and through only a small angle before re-

covering. At the same time controllability more balance has been added to the elevator and the elevator control has been simplified demanding a number of pulls in order to secure same excessive action.

The height of the vertical tail has been increased thereby increasing back the aileron trim and the area of the vertical surfaces and nose balance has been added to the rudders. The result is that control with one engine dead is perfectly in step with two engines operating. The airplane may be flown hands and feet off with one engine

dead and in case of engine failure the ship has almost no tendency to reverse toward the dead engine.

The performance figures on the Model 185 are as follows:

Cruising speed at 11900 feet and 7500 gross	225 mph
Landing speed at sea level	38 mph
Climb at sea level	1300 ft./min.
Take-off distance at sea level	300 ft.
Landing distance at sea level	325 ft.
Single engine ceiling	12,300 ft. (36 ft./sec. climb)
Service ceiling	22,000 ft.
Gross weight	7,500 lbs.
Empty weight	5,000 lbs.

Rearwin Cloudster



Latest addition to the Rearwin family is the "Cloudster" lightweight single-engine aircraft. Powered with either the 50 or 100 hp. Enviro-Turn engine, the Cloudster is constructed as Model 100 or 101 respectively. The 50 hp. model cruises at 100 mph and has a range of 375 mi., while the 100 hp. plane cruises 125 mph and has a range of 450 miles.

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August, 1939
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ARADO

Ar 96



SCHOOL, TRAINING AND LIGHT FIRST-LINE AIRCRAFT WITH MANY APPLICATIONS. ENGINE: ARGUS AS 10C, 240HP OR ARGUS AS 410, 450HP

ARADO FLUGZEUGWERKE G. M. B. H., POTSDAM-BABELSBERG (GERMANY)

RADIO

Dialing the Air Waves with Don Fink



5-Tube Receiver

Air Radio and Instrument's new outfit has a clip-on battery case patent.

GENERAL ELECTRICITY considering the size and weight is the achievement shared for the type AIL-AUX vacuum receiver recently announced by Air Radio and Instrument Co., Chicago Municipal Airport. Containing five tubes in a superheterodyne circuit (one tube in a tuned π -d circuit), covering the 100 to 410 kc band, and battery operated, the new receiver weighs but 12 pounds, complete with headphones and antenna. The overall dimensions are 12 by 6 by 7 inches. For installation in a small shop, the battery compartment may be separated from the receiver proper and connected by a cable. The list price is \$50.



Air Radio's 5-tube Receiver with Clip-On Battery Compartment.

Aeronautic Radio

A New Manual for Operators and Pilots by Myron S. Eddy, Published by the Randell Press.

ONE of the first books to devote itself exclusively to the problems of radio applied to aviation is "Aeronautic Radio" by Lieutenant M. S. Eddy, U. S. Navy Reserve, instructor in aviation radio at the Warner Technical School. According to the preface, the book is intended for all persons interested in any phase of the subject, including students, aviators, aviators, pilots and air transport operators. The treatment is practical, and the mathematics held to a minimum. The first four chapters are concerned with fundamentals of electricity as applied to radio, including the elements of circuit theory. Following are chapters on radio tubes, vacuum tubes and power supply. This information is, of course, available in many other texts, but has been included in order to make the book complete. The remaining four chapters are concerned directly with equipment used in aeronautical radio, including radio range speakers, tele-



Aeronautic Radio, Model D-5, receiver.

graph and telephone transmitters, converters, direction finders, instrument landing systems and radio traffic control. The last three chapters discuss the important problems of the installation and maintenance of aerial radio equipment. The book contains 302 pages and sells for \$4.90.

Four radio tubes are used very little in devices in the book, except the available fact that no book as general as this can go into detail on

all subjects, so make how interesting they may be to potential readers. If the reader must rely on a single book to obtain his knowledge of radio as aviation, this book certainly seems to be the best available. On the other hand it would seem desirable to consult some other volume, especially devoted to radio (for example, "Principles of Radio," by Kirtley Sawyer) which goes into more detail on the matters covered in the first chapters of "Aeronautic Radio."

At the end of each chapter there is a series of questions, over 200 in all, of a qualitative nature. Indications are given at the page on which the answers to these questions can be found. Lack of mathematical treatment, which would be serious in a book intended for design engineers, is not detrimental in a book intended primarily for operators who must merely understand the equipment designed by others.

All-Purpose Outfit

Aerovox's semi-receiver combination weighs but 18 pounds, has 8 watts carrier output.

AMONG the present crop of compact light-weight aviation radio equipment for aircraft, one of the latest to be announced is a neat little outfit made by Aerovox, Inc. of Laing, Long Island. The transmitter and receiver are available separately, but are built to operate as a unit. The receiver, model R, is designed for dry battery, or 6- or 12-volt storage battery (dismountable) operation, and covers ten frequency ranges from 350 to 420 kc and 2000 to 4500 kc. The controls include a toggle switch for shifting the frequency band, a volume control, tuning control, phase jack and antenna post. The weight of the receiver, less accessories, is extremely small, 3 pounds, 4 ounces. Equipped with headphones, power supply and cables, the weight is 13 pounds, 52 ounces.

The transmitter, model D-4, is a two-tube crystal-controlled outfit working on any of the standard frequencies. The carrier output of 8 watts can be 100 per cent modulated and the power drawn from a 12-volt supply is 475 watts. Its weight less accessories is 3 pounds, 8 ounces. The transmitter comes equipped with automatic microphone, aerial, dynamometer, ruber, etc. The combination of the type 12-4 receiver and D-4 transmitter weighs 18 pounds, including all necessary accessories for 12-volt operation, and costs \$210.00.



BUYER'S LOG BOOK

What's New in Accessories, Materials, Supplies, and Equipment

For the retreating of aircraft engine cylinders on the honing of new aircraft engine cylinders, the Bures Drill Co. Model 306-H hydraulic honer, manufactured by the Bures Drill Co., Bedford, Ill., has been widely adopted by the U. S. Army Air Corps and is also available commercially. The Model 306-H honer is a combined bore and straight line lapping machine and can be used for either type operation instantly. The machine is provided with a positive depth stop for blind hole honing so that the stress cannot be damaged against the cylinder bore, yet honing is performed the full length of the cylinder wall. The honing stones may easily be replaced with lapping blocks for a straight line lapping operation. The machine has considerable versatility for either retreating and repositioning continuously, or for repositioning only with hand or automatic indexing at the end of each cycle of repositioning.—*AVIATION*, August, 1939.

A line of gas designed steel aircraft hangars is offered by Elmer-Knox Company, Pittsburgh, Penn., through its standard building department. Delivery terms of these works or lot can be obtained. The standard line will include single-gable, double-gable up to 100 ft., but larger ones can be accommodated. Features include extreme weather-tightness, great resistance to corrosion, rugged construction, and quick erection, or dismantling for extension.—*AVIATION*, August, 1939.

Aircraft manufacturing frequently suffers from excessive tool and jig costs. To alleviate this condition it is sometimes possible to adapt a standard low-cost jig to a variety of machine operations, or for handling a variety of small parts. Such a jig is the "Mop" pump jig offered by the Elm Engineering and Sales, Inc., Detroit, Mich. This jig weighs less than 10 lb. and has working areas ranging from 1½ x 2½ in. to 5 x 8 in. In operation the Mop jig provides for special work holders on base, and lower side of top plate. Dual bushings are placed in the top plate when used as a drill jig. Work is clamped in place by operation of the pump arm, with an block handle handle, and is quickly released through operation of this changing arm. It is claimed that quick loading and unloading time are achieved, with lower operator fatigue, and the advantage of using a low-cost standard jig unit.—*AVIATION*, August, 1939.

For cleaning metal parts of such shape that they may trap solvent, chips, shavings, etc., a special rotating basket degreasing machine has been developed by the Detroit Box Products Company, Detroit, Michigan. Work comes from the solvent degreaser clean, warm, and dry, free from any trace of solvent or solvent vapors, chips or shavings. Careful design has effected natural circulation in cleaning solvent used. A special automatic elevator mechanism lowers and raises the rotating work baskets with minimum in the tank of solvent. A mechanism like is used to keep the rotating chamber free of insoluble material, so that the work is always immersed in clean solvent. The Detroit degreaser is manufactured in various sizes ranging in capacity up to two tons per hour.—*AVIATION*, August, 1939.

A new low-voltage night landing lighting system has been worked out for airport runways by the Westinghouse Electric & Manufacturing Company, for use in connection with instrument landings. Through use of the new lighting installation it is possible for pilots to make normal three point landings instead of the high speed approaches which have been used under low visibility conditions. The new Westinghouse system provides three green incandescent lights fixed with the ground, which indicate the beginning of the runway. Then, at 900-foot intervals, the amber-yellow quality of sodium contact lights, spaced on opposite sides, signals that the pilot is over the first 1,800 ft. of the runway. The last 3,000 ft. of the runway is distinguished by white incandescent contact lights and the far end boundary by three more green incandescent lights. Field tests at Akron, Ohio, have shown that the



Bures Hydraulic Honer



Elmer-Knox Hangar



Detroit Service Degreaser



Elm Mop Drill Jig



Westinghouse sodium contact light



Bures Exhaust Gas Analyzer



Dwyer Vaporometer

redmen contact lights are visible at a distance of 400 ft., when conditions are such that incandescent lights are visible from only 20 ft. All the light from the four contact lights is directed at an angle of four feet to two degrees, which is in the direction of the eyes of the pilot in the incoming plane.—*AVIATION*, August, 1939.

Airline passengers may now ride on air in more ways than one, at least when patronizing Transcontinental Central Airlines, which has just introduced its office floor of glass with seats upholstered in Avirex, the new seared linen product developed by The Prunette Tire and Rubber Company. Avirex has seared advantages which make it ideal for use as a cushioning material, an aircraft seat. It is light in weight, compact, and durable. A pressure cushioning effect is provided by millions of inter-connected air cells acting as sensitive shock valves and thus permitting or rejecting within the cushion to compress or escape slowly when pressure is applied; i.e., when a fat man sits down. Upon removal of the fat man, or pressure, the material immediately snaps back to its original form and restores.—*AVIATION*, August, 1939.

To determine the fuel to air ratio of an engine or set of engines in an aircraft in flight in efficient instruments has been announced by the Brown Corporation, of Norwood, N. J. Known as the Brown Exhaust Gas Analyzer, the instrument consists of a unit (one for each engine) linked to the flywheel, and an indicating instrument mounted in the cockpit. The theory of operation is that electrical instrument senses the pressure of carbon dioxide, indicating a lean mixture, and hydrogen, indicating a rich mixture. The movement on the panel shows the pilot when his engine is running at lowest mixture without danger of stalling.—*AVIATION*, August, 1939.

Detergency designs dark in the case of a unique cleaning solvent developed by the Curran Corporation, Middle, Mass., for use in cleaning small metal parts. The new dark detergent has the highest performance of benzene and leaves a light phenolic film which acts as a rust preventative measure. According to the Curran development laboratory the new solvent has a clear sparkling appearance (B.D. before use), and is highly penetrating, and inflammable. Small metal parts dunked in the solvent assume a mirror appearance without etching, marking, or loss of weight, and such parts may be rinsed with water to produce a chemically clean surface where painting or plating is to follow.—*AVIATION*, August, 1939.

Cleaning solvent fumes is a difficult chore under ordinary circumstances, but the job has been simplified greatly by the use of a cleaner developed by the Magrant Chemical Company of Greenwood, N. J. Magrant's Cleaner is available in two specifications—62 P.S. and 62 P. The fumes is intended for cleanup on old fumes or for severely difficult jobs while the latter is for normal work. Although some report that simple fuming is usually sufficient to remove the grease and dirt after the cleaner has been applied, the manufacturer has developed a 3-ounce brush which incorporates a scraper and squeegee in the brush.—*AVIATION*, August, 1939.



Dwyer Vaporometer in the E.A.L. line of Dwyer Vaporometer

4

Why "just get by" with a makeshift bench when you can have a "HALLOWELL" STEEL WORK-BENCH at so little cost?



Fig. 232.—"Snap-on" Steel
Work Bench.
Design is new.

"The [new] Bantams have lost 2 key features needed for Aviation Production and Maintenance Work."

1-SMOOTH WORKING SURFACE

"McBowell" Beach Tops are one piece of sturdy steel, perfectly flush ... and they stay smooth ... won't get all scuffed ... won't splinter or warp.

2-LASTING RIGIDITY

Flanged steel leg construction allows pivoting
slidably with a sliding marking surface.

1-EASY MOVABILITY

Whenever changed requirements demand, "Rollaway" Benches can be taken apart, moved to a new location and set up again with no effort. Be unlike the leading wood that results when you try to move a wooden bench. Two is any number of "Rollaway" Benches can be joined and to end to create a continuous smooth working surface.

—TWO OF THE TEN COMBINATIONS AVAILABLE



Fig. 4B — "Rollerball" shoe with thick sole, a removable top of knitted material and the thick sole work.

Fig. 4C — "Rollerball" shoe with thick sole, a knitted sole in front for the thick sole work.

Send now for the coloring that tells about the cam:
where home

STANDARD PRESSED STEEL CO.
 BRIDGE JAMUNTOVN, PENNA. BRIDGE
 BRIDGE BRIDGE BRIDGE
 BRIDGE BRIDGE BRIDGE
 BRIDGE BRIDGE BRIDGE

DEFENSE

Congress Comes Through

By Minnie Stubbins

The Seventy-ninth Congress saw the start of '78 in the reading. This country and the world moved its air forces way out in front as first line weapons. The serious makers put over \$30,000,000,000, mostly 100-day paper, into these wars for 1948! The pre-war figure included \$1,000,000,000 worth of Roosevelt money to defend at home and hemisphere. And aviation, including car and military flying tanks with all appearances thereto, got nearly \$700,000,000, or 23 per cent of the total sheet.

Three antitrust projects are being financed in the third de Sweeney bill—a congressional device for paying off agencies that run short of change in getting the appropriations made: (1) the CEA antitrust program costing \$8,870,000 this year; (2) about 30 million dollars for road pay to the Post office on new domestic routes; and \$1,418,500 for foreign mail pay, which amount was diverted from \$300,000 because transatlantic service got started the first of the fiscal year instead of later, as expected. (U.S. SEN. JAMES EASTLAND)

As the writer of a bill aimed directed for passage authorizing the Secretary of War to dispose with the "one-hundred percent" of competition bidding on construction projects outside the United States, I was asked primarily to support most world and anti-sect defense of the Panama Canal for which \$12,715,000 was authorized under the expansion act. The work would be done by contractors on a cost-plus fixed-fee basis, with profits limited to ten per cent. Of course the law, if passed, would apply to all different air and other bases, and would run into the question especially of competition with the present Panama Canal operation, which will be made its exception.

Washington seemed that the Army was talking about its new streamlined air-cushioned engine primarily when it announced the beginning of a fleet of cargo-lifting ships better than those of any foreign power. Those contracts had already been let among them in small, speed-urgent tenders, and orders for bombers and heavier equipment would follow. Within about two and a half years, Army was

Navy expect to have on the line most of the 5000 planes now authorized, not less than 5000 for Navy and not more than 6000 for Army.

Washington Conference

War Dept. talks contract system with industry

War Department and Air Corps officers and officials of manufacturers of American-made airplanes and aircraft factories in Washington on July 25, to discuss problems which had arisen in connection with the Air Corps expansion period. After the conference, Lt. Gen. Hensley, Assistant Secretary of War, reported that many comments had been made orally at the meeting and that he was more assured that the industry would be able to meet the demands of the war.

Mr. further details of the conference were made public but it is generally understood that considerable discussion took place of suggestions to permit the War Department to place aircraft contracts without competitive bidding. Such permission, it is thought, would help speed the whole program and render it much easier for the War Department to take an

manage all the facilities of the smaller manufacturers. Congressman Andrew M. May, Chairman of the House Military Affairs Committee, recently introduced a bill permitting such contracts.

Bill for Pilots' Pool Becomes Law

C.A.A. to trade 95,000 in five years

The bill authorizing the C.A.R.'s program to train civilian pilots was signed June 27, by President Eisenhower at Hyde Park, N. Y.

In its final form the bill provides for the training of 15,000 pilots in the next twelve months.

and authorized a \$3,874,000 appropriation for the purpose. It authorized an additional \$1,000,000 appropriation annually until mid-1964 for the training of another 20,000 scientists each year.

Between 300 and 400 universities and colleges are ready to offer the introductory "ground school" course when the next academic year begins in September.

**CURTISS
PURSUIT AIRPLANES
of the
P-36 TYPE**

Since originally developing the design, Carlin has built more than 500 airplanes of the P-30 type, which is now used as the standard pursuit plane of the U. S. Army Air Corps.

It is with extreme pride that this announcement is made in the operation of this stepdown by many we locate throughout the world with Carbon in the foreman builder of parent aircraft.

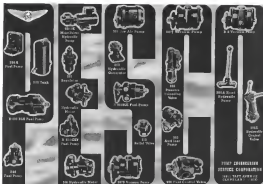
The major portion of production work at West Coast's Curtiss plant has been done during 1949 and Curtiss factory facilities are geared to continue the manufacture of many additional units.

Planes of the P-35 type are extremely fast and maneuverable and may be equipped with various combinations of armament to make them unusually effective offensive or defensive units of modern aerial warfare.

CURTISS AIRCRAFT DIVISION
CURTISS-WRIGHT CORPORATION
Buffalo New York

Everitless

**CONGRATULATES The U.S. ARMY AIR CORPS
on 30 YEARS OF PROGRESS**



WHEREVER YOU FIND OUTSTANDING AIR ACHIEVEMENTS PESCO PRODUCTS ARE PLAYING THEIR PART

An airline wins a maintenance award... PESCO standard equipment is a factor. A racing plane sets new speed records... PESCO pumps operated faultlessly to make it possible. Higher and finer performance standards are set by military aircraft... PESCO products help raise them. Eight airlines are cited by the National Safety Council for outstanding safety records... all of them standardize on PESCO equipment for unflinching service.

CONTRACTORS TO THE U.S. GOVERNMENT
FOR APPROVED AIRCRAFT ACCESSORIES

PUMP ENGINEERING SERVICE CORP.

12910 TAFT AVENUE, CLEVELAND, OHIO
DIVISION BORD-WARNER CORPORATION

AVIATION MANUFACTURING

Douglas Rushing DC-5's Delivered to Start this Fall

Production of twelve Douglas DC-5 transport, the latest Douglas economical model, is under way at the St. Louis Division of the Douglas Company. Deliveries to domestic and foreign airlines are scheduled for the latter part of 1935 and thereafter. Among these slated to receive DC-5s are KLM, which has ordered four; Pennsylvania Central Airlines, which has placed an initial order for six; and Southern of Birmingham, Columbia, Tenn., of 12 passenger capacity, the DC-5 carries a crew of three. The value is divided into three long-range configurations. The high wing, four-engine and one of a tandem type landing gear help make the DC-5 well for small field operation. Wing span is 78 ft. 2 in., gross weight 9000 lbs., and top speed 200 m.p.h. Engineering and technical subsidiaries of the DC-5s were developed under the supervision of Edward H. Rasmussen, chief engineer of the St. Louis Division, and Richard B. Dink, its general manager and a well-known of the Douglas Company.

A vintage ceremony was held when the first Douglas DC-5 on June 19, when forty six, most of these present in the aircraft industry, landed at Douglas in the St. Louis commercial aviation. Features of the ceremony were the setting and service of a 75 pound cake. On hand were Donald W. Douglas, his father William A. Douglas, and other Douglas company executives. The DC-5 was piloted on the anniversary flight by Major Carl Cover, senior vice president of the Douglas company.

Mead Assigns from United Aircraft

United Aircraft Corporation has work assigned with regard to the construction of George F. Mead as vice-president, engineer, and chief engineer of the corporation.

Mr. Mead, attended the Massachusetts Institute of Technology. His activities in the aircraft industry began in 1917 when he was associated with the Wright-Martin Aircraft Corporation at New Britain,

Conn. Mead, an experimental engineer, in 1924, he left the position to become manager in charge of the power plant laboratory of the United States Army Air Corps at Dayton, and in 1926 was made chief engineer of the Wright Aeronautical Corporation at Paterson, New Jersey. He resigned in 1930 to become one of the co-founders of the Frank & Whitney Aircraft Company of which he was vice-president and chief engineer. He was elected vice-president, director, and chief engineer of United Aircraft Corporation in 1930, and in 1931 was made vice-president, director, and chief engineer of United Aircraft Corporation. In 1937, Mr. Mead was awarded an honorary degree of Doctor of Science by Trinity College.

Northern Offers Shares

Northern Aircraft, Inc., new vehicle for the engineering and design talents of John E. Northing, was off to a flying start late in June with the offering of 400,000 shares of common stock following approval of the Northern registration statement by the Securities and Exchange Commission in Washington. Northern announced that a factory would be erected immediately on a 15-acre site adjoining Long Beach Municipal Airport. The factory should be in operation within 60 to 120 days, be stated, and will undertake with connecting work for other aircraft firms while preparing its own design. Ap-



TWO IN ONE! The turbine power plant of Vega's new Starliner.

proximately \$2,000,000 in losses caused by the initial financing, of which about \$500,000 will be received in January and mid-May, and the balance held in reserve or working capital. Preliminary personnel of between one and two thousand persons is expected to be required in the near future. Associated with Northern are Laurence T. Cook, chairman of the board and general manager, George Irving, vice president and mechanical superintendent, E. A. Schmitt, sales and test pilot, and Raymond J. Medford, treasurer and comptroller. Activity of the company of the time will be confined to the design and

production of small and medium size military aircraft of patrol, attack or light bomber type.

Vega Tests New Ship It "exceeds expectations"

Successful completion of first flight tests of the new Vega Starliner has been announced by Max Stern, president of the Vega Aircraft Company, a subsidiary of the Lockheed Aircraft Corp., Vernon, Calif. Vega test pilot on initial flight, reported that flight performance of the new model exceeds design expectations. The Starliner is now to be subjected to a 100-hour schedule of flight tests preliminary to the issuance of its AEC from the CAA.

An unusual source of interest has been evidenced in the progress of the Vega Starliner, due to the many unusual features incorporated in the design. Of particular, but unconventional appearance, the Starliner is an all-metal low wing monoplane with retractable landing gear, accommodates 100 passengers, and top to the passengers. Landing gear is of tractor, or "beveling" type, and is retractable vertically, leaving part of each wheel exposed for emergency fully landing. Most unique feature of the plane is the use of the new Monocraft fuselage powerplant, incorporated two Monocraft CES 100's by engine design a single Monocraft-Standard constant-speed



HAPPY RETURNS: William E. Douglas congratulates his son Donald on a 25th anniversary.



THE NEW HAMILTON/Designed the PC-8 Trainer, this two-place biplane mounts a Warner Super Sirocco

propeller through over-riding blades. The over-riding blades eliminate drag of wing sections, and provide extra-long engine life.

The Sirocco has a gross weight of 4,000 lbs., top speed of 220 mph, cruising speed 190 mph, cruising range of 1,400 miles, and a ceiling of 21,500 ft. Landing speed is 51 mph.

Our Vee Standard has been purchased by Red Canyon Airport, at Kansas City, and will be put in service during this summer on the North and South Texas part of the line. A preliminary production lot of five Vee Standard is now coming through the factory.

Hamilton Goes to Town Tests trainer; starts production; names distributors

Already in commercial production, the first plane delivery, the Warner Aircraft Company conducted test flights on its new two-place trainer early in July. The new trainer is being offered at the lowest of its kind and primary flight trainer, providing instruction in operation of flap, retractable landing gear, retractable propeller, and flight in a plane of type and performance desired by high speed military transport.

Production is now under way at a \$24,000 order from the Civil Aeronautics Authority for a fleet of four-place Hamilton extra monoplanes to be used for personnel transportation. This is an all metal low wing cabin monoplane powered with the Warner Super Sirocco engine of 140 h.p., and equipped with flap, retractable landing gear, and retractable propeller.

Hamilton distributors recently appointed include Jack O'Hara for Minnesota, North and South Dakota, and Northern Wisconsin, Howard Adler for New York, E. W. Wallace for New England, Harry Hearnish of Austin, Texas, for the South West, Frank Clark for North and California and Nevada, and John F. Fafnir for Southern California and Arizona. The Hamilton PC-8, two-place trainer plane, has already been given approval by ATC, and the PC-10, a four place model of similar type is now undergoing tests.

The new Sirocco, designated the PC-8, is now undergoing preliminary flight tests which aim, to date, to demonstrate performance exceeding design specifications. The PC-8 Trainer already assembled. The Sirocco powered transport four place plane recently completed its inspection. It is quite similar to the Army's current all-metal low wing patrol plane. Landing arrangement of the trainer is a two wheel landing gear, a smoothly rounded cabin incorporated in the design are features which make it suitable for tracking light vehicles from primary through advanced flight instruction. Although the plane is equipped with electrically operated flap, retractable landing gear, and retractable propeller, it is a thoroughly practical primary flight trainer without any of these accessories. Following sufficient instruction in operation of the plane as a conventional trainer the student is introduced step by step to the operation of flap, retractable landing gear, and the retractable propeller.

The Warner PC-8 Trainer has a wingspan of 31 ft. 10 in., width of 20 ft. 6 in., height of 7 ft. 8 in. It is equipped with a Warner Super Sirocco engine rated at 140 hp.

Re

*The Right Fafnir
Recommended on the Training
Board for Complete Security*

*Easy
Cure for*

"PULLEY PALPITATION"

It took no long time of development to produce the Fafnir Pulley Bearings that soothe wobble and doctor in the Lockheed big game show case. Fafnir design, materials, and precision manufacturing provide the frictionless action, combined with extreme rigidity, that makes a "wobble" control system—through constant flying hours of test and cold and warm weather. And on the ground, give normally non-wearing bearings eliminate the maintenance that goes with wobble, sloppy cables, frequent tightening of turnbuckles. For your application, you'll find the right bearing, with complete specifications and experience in Fafnir Catalog, which will be sent to you promptly on request. The Fafnir Bearing Company, Aircraft Division, New Britain, Conn.

FAFNIR

THE BALANCED LINE . . .

50,000 RESULTS OF TEAMWORK {Fafnir and Hamilton Standard}



"Grief" makes sense. That's why the student pilot who would never in October ever page one of your newspaper, while the story of the fellow flying to midlands will go back under the ropes for corn bread.

That's why, to illustrate teamwork, we tell you about the education process that led up to

teamwork inside special bearings, instead of suggesting them. We like to find aircraft manufacturers who think they need "improvements" and introduce them to standard Fafnir already made for the job. But those men's words, so many of the Hamilton Standard says.

The bearing wanted was the one at the end of each blade, as which the blade runs when it changes pitch. On a 3100 h.p. engine, this bearing carries an 11-ton thrust load, you must permit the blade to turn freely.

First try was to install two halves of a split bearing on the finished propeller blade. Vendor satisfied, but experience when precision must be maintained. Second try, a one-piece two-piece thrust bearing slipped over the end of the blade and secured by a female. Vendor OK, but the second one female might not "take it." Final result, a finished

bearing that's slipped over the end of the finished blade. Made of special steel, the bearing withstands a temperature of 910°F going through the blade heat treatment. On the finished blade, it's right in place, ready to go 50,000 such bearings now going unobtrusively.



Proof that in Fafnir's separate Aircraft Sales and Engineering department, we make it our job to solve bearing problems through cooperation. Teamwork that produces intricate special bearings when necessary, but usually finds the right bearing for every job on every day properly. From the most complex job in America, The Fafnir Bearing Co., Aircraft Division, New Britain, Conn.



50,000 Fafnir Bearings now carrying up to 51-ton thrust loads in Hamilton Standard Propellers the world over. Most of the time,

Ball Bearings

... MOST COMPLETE IN AMERICA

For Aircraft
Engines and
Controls

THE Kollsman Company has been serving the Army Air Corps since 1928, the year the company was founded. To-day, Kollsman Instruments, developed for or in cooperation with the Army Air Corps, play an important part on every army instrument board. Examples are Kollsman Sensitive Altimeters, the present Air Corps standard; Kollsman Electrically Heated Pitot-Static Tubes; Kollsman Vertical Speed Indicators; Kollsman Manifold Pressure Gages of the capsule type; special types of accelerometers, and several types of magnetic compasses. Kollsman Rim Lighting, now used by many manufacturers of instruments for the Air Corps, is another outstanding Kollsman development.

The Kollsman Company salutes the Army Air Corps in recognition of its constant encouragement of greater precision, greater reliability, greater serviceability in aircraft instruments.

KOLLSMAN INSTRUMENT CO., Inc.

8008 FORTY-FIFTH AVENUE ELMHURST, NEW YORK
WESTERN BRANCH: GRAND CENTRAL AIR TERMINAL, GLENDALE, CALIFORNIA

SEE THE ARMY AIR CORPS EXHIBIT AT THE AVIATION BUILDING, NEW YORK WORLD'S FAIR

ENGINEERING

Build New Chief at Boeing Headed Commercial

Wellwood E. Beall has been promoted to the post of Chief Engineer of the Boeing Aircraft Company. Formerly Beall was engineer in charge of all commercial projects.

Beall studied mechanical engineering at the University of Colorado, then attended the Guggenheim School of Aeronautics at New York University, where he received the degree of Bachelor of Science in Mechanical Engineering and Master of Science in Aeronautical Engineering.

After a period of aviation and aeronautical design work in California, Beall in 1926 became an instructor at the Boeing School of Aeronautics in Oakland, California, which was then affiliated with the Boeing Aircraft Company. He was subsequently placed in charge of all engineering at the school. In 1931 he became a sales engineer for Boeing, then was made for eastern representation in which capacity he spent a year in the Orient. Upon his return he was transferred from sales to engineering work, and in the summer of 1933 headed chief commercial projects department.

"Engineering Stylist"

William S. Laidlaw, former member of press director of engineering styling and passenger comfort problems for Douglas Aircraft Company, is

interviewed with development of the DC-4, DC-6 and other designs, but created a new environment for styling problems of the new Lockheed "Stargazer" four-engine transport. He is also active in styling the Harvard Aircraft Co. "Aeromarine Engineering Stylist," according to Laidlaw, operations in operations of lighting, ventilation, seat arrangement, and other engineering, and other transportation. He is active in problems concerned with passenger comfort.

New Sleeve-Valve Engine

The single-cylinder sleeve engine is not completely new at Allison. Recently Allison Cooper Beall, former member of the engineering staff of the Ford Truck Company, the Wright Aeronautical Corporation, and the Martin Manufacturing Company, moved up

date point of development work by partially demounting a single-cylinder, single-sleeve-valve, six-cylinder engine which he pronounced a "joint test and ready for application to aircraft propulsion, or in general auxiliary power units. The Beall test engine is a four-cylinder cycle engine similar in general to the well-known English Bristol type. The sleeve valve operates directly in a cast aluminum block. The test engine has been driven at speeds

up to 3,500 rpm and has been operated for more than 140 hrs at speeds approximately 3,000 rpm. Beall has just set perfect plans for commercial production, but indicates he has solved the engineering problems involved in production.

Alfonso On Stratiellers

Fieldwork continues on the engineering tests of the Boeing No. 2 Stratieller June 22 by Edward E. Alfano, Chief Test Pilot. "At an altitude of 10,000 feet, as far as atmospheric pressure in the engine was concerned, we were over 10,000 feet," Alfano is making a descent from 14,000 feet to 8,000 feet. A mile and a half, but altitude was held, but there was no sensation of rapid descent because the pressure regulating device kept the cabin atmosphere at virtually a constant pressure, as though in level flight. As for the Stratieller, Alfano said, "The instruments showed it was one degree below zero. Contrarily, the engine was not. Fortunately, outside, and inside the cabin we were perfectly comfortable in our air sleeves."

The altitude-conditioning system controls merely of modulation and blowers that increase the pressure of the air as it enters the plane's conditioning system, and automatic control valve that regulate the intake atmosphere pressure so as to be level constant. Two air units are installed in the Stratieller, either one of which is capable of handling the entire cabin pressure requirements alone. Reported tests of the oxygen-charged cabin, which was made since the initial test, the last one reported being July 5 when the ship was taken to 25,000 feet. Even at this altitude, according to Edie Alfano, the pressure relieving equipment "functioned perfectly," and no difficulties inside the cabin were as if flying at ordinary altitudes.

In case you don't remember the details, the Lockheed test cabin operates across approximately 10 per cent of the altitude. They are not continuous, being regulated in five increments. The test cabin is a stepping structure in the wing structure. Openings start in the middle of the joint, but in the closed where standard door level installation bore of 10-degree normal angle at altitude, high speed motion, or vibration, there is practically no flow of air through the slots. As a result there is no appreciable increase in the drag of the wing at cruising speeds. Even at top speed, the slots cost the Lockheed only two mph



ENGINEERING A MYSTERY: Last month our New Department, directed a new flight test pilot, Mr. W. S. Laidlaw, who was recently assigned to the result of aircraft work by A. E. W. Laidlaw (top) and Laidlaw and Alfano.



WELLWOOD E. BEALL
Alfonso Stratieller in his position

...Today!

The existing new Bellanca 149 is ready for the airports. By the outstanding 3-place cabin job—select the leading jet—lead along for a minute or more in the comfort of the ultra-quiet 3-place cabin—separate the safety and pilot's value of Bellanca leadership.

It's a BELLANCHCA!

Which for the 149 at your local airport, flyway price \$3975 with retracting gear, hydraulic landing, 50 g.p.h. rated engine * * *

BELLANCA AIRCRAFT CORP.
New Castle, Delaware

961613

Pan Am Into L.A.?

Following shutouting of the north of Pan American Airways' new Boeing Clipper as the American Clipper, a derivative of Los Angeles harbor, it was announced by Mark T. Wilson, Pan-American director, that Los Angeles will be included in a regular port of call on the U. S.-New Zealand service for which a franchise to operate on a fortnightly basis is being requested of the C.A.A. The present plan is to operate the American Clipper between Boston and New Zealand. No indication has been given as to how Los Angeles will be included in future Pacific flight programs.

T.W.A. Names Treasurer

E. Lee Tolman, 34, of Chicago, has been appointed Treasurer of Transcontinental and Western Air, Inc. with headquarters in Kansas City, Mo. He succeeds Frank C. Wilson, formerly Vice President, Treasurer and Secretary, who resigned as Treasurer but remains as Vice President and Secretary.

In assuming his new post, Tolman brings with him six years of aviation experience with the Boeing Aviation Corporation, having served as Assistant Treasurer of that company from 1935 until 1939. After graduation from Northwestern University in 1936 he was associated with the J. E. Jacobs and Company, Chicago, until 1939.

In joining T. W. A., Tolman leaves the presidency of Tolman Travel Supplies, Inc., affiliated company of Gateway Mills of Los Angeles, Cal., one of the largest outfitting firms in the United States which he joined in 1931.

TRAFFIC

✈ American Airlines carried 834,812 passengers during June, representing an increase of 22 percent over June 1938. A new all-time record of 288,384 passengers was made during the first 6 months of 1939, a gain of 50 per cent over the corresponding period in 1938.

American's June passenger-increase total also set a new world record—18,327,303 passengers—made compared with a total of 18,042,000 during 1938, an improvement of 94 per cent.

✈ British Airways Traffic over South America during June reached a new high per month

AIR
TRANSPORT
INDICATOR

July 1, 1939

157.8

Which is the rate of the passenger miles reported by the Air Transport Association for the month ends for June, 1939 to the figures for June, 1938.

Needs for the passenger miles for the month ends for June, 1939 to the figures for June, 1938.

✈ Flying 1,226,000 passenger miles. This represents an increase of 88.86 per cent over June 1938.

✈ Canadian Colonial Airways carried a total of 1,011 passengers during the month of June, a big gain over the 250 passengers for June 1938. 1,011 passengers were carried for the first six months of 1939 against 6,202 for all of 1938.

✈ Chicago and Southern Air Lines for a total of 129,415 passengers during June, representing a 39.2 per cent increase over June 1938.

✈ Great revenue from the nation-wide air express service of the Standard Express Agency for May increased 20 per cent over May, 1938. The agency reported today. This monthly revenue record has been recorded only three times since the service was inaugurated in 1929. May is not ordinarily a heavy month in air express.

✈ Trans-Canada Air Lines reports 303,510 passengers during June, representing an increase of 17 per cent over the preceding month. A total of 1,805,490 passenger miles has been flown for the first five months of the year.

✈ TWA reports a total of 2,320,100 passenger-miles flown during the month of June. This represents an increase of 11.35 per cent over 1938 figures.

✈ United Air Lines reported an unofficial total of 15,860,000 passenger miles for June, which is a 21 per cent increase over June 1938. This is a new record high for June month, the previous high being 15,125,000 passenger-miles for August 1937.

THE
CONTROL TOWER

by DAN LARSEN



It's safety makes the more go

New York's Grand Central Terminal for the air line companies is finally getting through. Details have been somewhat changed since the last of our preliminary announcements. But the building is to be a new, an addition from the control of the old Grand Central Terminal and the old Grand Central Terminal.

Accommodations will be very large. As you might know, the old Grand Central is at somewhat higher level than Grand St. Also Grand St. freeways is built locally at a very high premium. As a result, the old St. freeways will be occupied by roads which will include a drive from a corner of the old Grand Central Terminal. As you might know, the old Grand Central is at somewhat higher level than Grand St. Also Grand St. freeways is built locally at a very high premium. As a result, the old St. freeways will be occupied by roads which will include a drive from a corner of the old Grand Central Terminal. As you might know, the old Grand Central is at somewhat higher level than Grand St. Also Grand St. freeways is built locally at a very high premium. As a result, the old St. freeways will be occupied by roads which will include a drive from a corner of the old Grand Central Terminal.

Such a terminal will be a big boost to the industry. With prospects of North Beach being built for us sometime this year (if the W. G. A. doesn't say so), the old Grand Central, it will be a big boost to the industry. With prospects of North Beach being built for us sometime this year (if the W. G. A. doesn't say so), the old Grand Central, it will be a big boost to the industry. With prospects of North Beach being built for us sometime this year (if the W. G. A. doesn't say so), the old Grand Central, it will be a big boost to the industry.

Terms of the deal, although not fully disclosed, are understood to be very favorable to our firm. One of the main reasons for the deal is that the old Grand Central is at somewhat higher level than Grand St. Also Grand St. freeways is built locally at a very high premium. As a result, the old St. freeways will be occupied by roads which will include a drive from a corner of the old Grand Central Terminal.

Traffic opens these days certainly more than before. Most newspaper announcements clearly credit the boom to the new World's Fair. And so it was. But the World's Fair has helped. One of the main attractions of the Fair is that the foundation work of the old hotel has been left in place and can be fully utilized for the new building. Hence the new building will be a big boost to the industry. With prospects of North Beach being built for us sometime this year (if the W. G. A. doesn't say so), the old Grand Central, it will be a big boost to the industry.

Airlines report on HYDROMATIC PERFORMANCE

During the past year Hamilton Standard and equipment on the major airlines of Hamilton Standard asked this question of

Hydraulic Propellers have become standard in North America. How have they performed? airline operators...and reports these answers:

ACCUMULATED SERVICE. Hydromatic Propellers have already had hundreds of thousands of flight service hours, and have flown more than 60 million propeller miles on domestic airlines.

"We now have 54 propellers in use. The oldest propeller has approximately 2700 hours at this time and 43 propellers, or nearly one-half, exceed 3000 hours. . . . The total time of all Hydramatic propellers in service on our air line at the present time is approximated nearly 170,000 hours."

We now have a fleet of 53 propellers in active service, including spares, a large number of which have over 2,000 hours total time. The highest time propellers are a little above 2,600 hours. This represents a total accumulated service time of over 125,000 hours. . .

— United Air Lines

Hydrotransit propellers in our operations have now aggregated a total of more than 60,700 service hours and some of the individual propellers have accumulated, as previously indicated, over 1,500 service hours. — *Transcontinental & Western Air*

Serphocampa clivosa pupellae survive best with thirty per cent.

WINTER PERFORMANCE. Hydromatic Propellers came through the season intact to extreme cold weather operation with flying colors.

¹ "We consider operation at functioning of these units has been reported which could be considered as low temperatures." — "Can stable to give heat ... an important as our dimensions calculated in terms of consistently reported upon pay ... ensure that the temperatures recorded were to degree ..." — *Trans-Canada Air Lines*

*Even in the coldest temperatures we have had no reported air space difficulties reported in the smaller dome, on the ground, or in flight. — American Airlines

"We have not had any trouble — even with temperatures 14 or 15 degrees below zero."

*In emergency situations last limited hours of operation of Hydrovacous propellers in temperatures ranging from 35 below to 120 above we have had no delay or income
— President J. D. Jones

MAINTENANCE. Full pressure lubrication and improved design of the Hydromatic Propeller has greatly decreased the maintenance requirements.

The maintenance costs for routine daily service have been reduced almost 95% in comparison with the old nonrefillable propellers. This is due to reduction in the grinding and servicing required and to the very marked improvement in synchronisation which has reduced the maintenance problems in use. Our average annual cost for the constant speed propeller was \$20.47 per overhaul, while on the Hydromatic propeller it has increased to \$3.66 per overhaul.

"During this entire period of operation we have found our requirements for implants gone to be gratifyingly small, which, of course, directly results in reduced maintenance cost, both during overhaul periods and daily average operations."

* Beyond very severe difficulties with increased drug plant rates replaced with solid plants and very slight wear on cam rollers no difficulties have been experienced with these assemblies and no major malfunctions have been noted. — Tracy Canada All Lines

"That seems to show heliports they are less expensive to maintain than constant speeds of the same size. Consider from Noble's perspective." — *New York Airlines*

GENERAL COMMENTS

"All in all, our experience with *lipidomics* has been most satisfactory. We consider the program thoroughly "worth" our effort, and they have proved to be exceptionally reliable and consistent in their operation. We feel that they have very definitely contributed to the safety of air transportation without any appreciable increase in the maintenance or cost of operating. May we congratulate you upon the production of such a thoroughly reliable machine?"

*The speed prevailing is substantially better than ever obtained with the common speed propellers. For instance, on a heavy ship a few days ago it was necessary to maintain the speed of 14 knots per hour at a 1000-horsepower shaft. — *Science*, 4th June 1907.

*I would like to emphasize the fact that these propellers have given us precisely what we have needed.¹ — *Trans Canada Air Lines*

*You will probably be interested to know that since the introduction of Hydrus, propellers on our DC-3s equipped for operating experience with them has been very satisfactory and flying personnel are particularly pleased with the accuracy of engine revolutions; this is maintained on these installations.

Here, then, are the reports . . . comprehensive, authoritative, conclusive. After a full year of rigorous service, a year in which these airlines have established amazing new operation records, unanimous tribute is paid to Hydromatic performance. Again Hamilton Standard, with the Hydromatic quick-fathering Propeller, plays a leading part in aeronautic progress.

HAMILTON STANDARD PROPELLERS

One of the finest divisions of
UNITED AIRCRAFT CORPORATION • EAST HARTFORD, CONNECTICUT

The following airlines
now have Hamilton
Standard Hydromatic
Propellers either in use
or on order.

AMERICAN AIRLINES
EASTERN AIR LINES
NORTHWEST AIRLINES
T. W. A.
UNITED AIR LINES
PAN AMERICAN
AIRWAYS
TRANS-CANADA
AIR LINES
CANADIAN COLONIAL
AIRWAYS

BRITISH AIRWAYS
IRISH AIR LINES

AIR FRANCE
SABENA

OLIVE AIRLINES CO.

L. M. ROYAL DUTCH
AIR LINES

K. N. I. L. M.

SWISS AIR

AFRIQUE ET ASIE
A. ASIE

SOUTH AFRICAN
AFTERLIFE

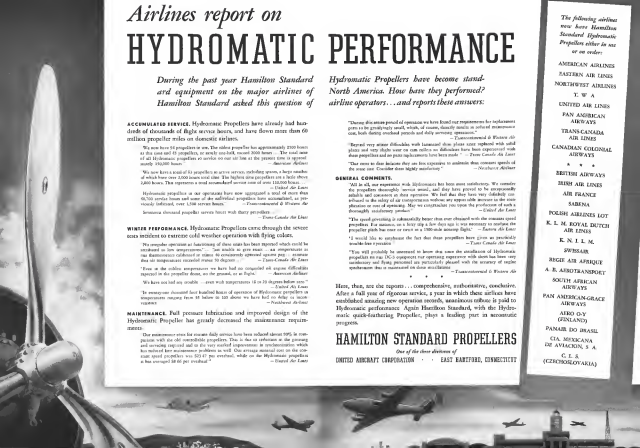
AMERICAN-GRACE

ALWAYS
ALWAYS

(FINLAND)

A. MEXICANA

AVIACION, S. A.



Vultee's Special Machines

(Continued from page 37)

stock of aluminum-silico, or other alloys of desired. These sections may be U, V, L, or hat shaped, however, other shapes or variations can also be formed. From these straight channels we roll rings of circular or oval section, or any variation. The machine is also used to roll rings from various extruded sections, or from channels, angles, or beams.

Another process of even wider application, and of great significance from a production standpoint, is the use of computer-aided machinery. This is the use of computers to solve problems of how best to cut out of large sheets from sheet stock. We shall see the process in context as we discuss the use of computers in rubber design and make our point as reinforced. There are two ways now in vogue in aircraft plants for the computerizing, and the computerizing, of a drawing. One is to use a kind of camera board mounted on a table, and move the work part the least by hand or by a motor. The other is to use a kind of gangle the entire drawing. When it is realized which we have developed and now use in the rubber shop. Another method provides for mounting the drawing on a table, and moving it along the work, which is held stationary on the table. In either case a lighted board is used. This is the use of computers in rubber design. This is the use of computers in the rubber world reader to chapter, except that special cutters are used and the use of computers is considered for the work of cutting in rubber design.

Profiling a Gun-Ring Mount

A good example of the application of profiling to a process machine operation is that of taking the outside stock diameter out on the expanded section governing means. This part comes in as an flat extruded stock of approximately I-beam cross section. We roll the flat stock into a segment of a circle and then mount it in a special fixture on the profiling table and pull the work past the cutting head by means of a hand-operated drum and cable. The result is a circle diameter out held to within 0.002

This part then goes on to the machine shop for milling and drilling operations which have been described previously. The same part, if machined on a lathe, would require an

unusually large among. Also, we cut little time, under the best of conditions, by at least 50 per cent. The machine used is adapted from a Glenn wood shaper with a 48-in. table top and a vertical spindle for the router, which rotates at 11,400 rpm. (See p. 14.)



The Tropic Press at San Francisco, California, is the world's largest press. It has a total working pressure of 1,000,000 lb. The total weight of the press and its foundation is 3,300,000 lb. Designed by North American Engineers it was built by Johnson Heavy Iron Works. For further data see page 75.

cally driven. A 1-in. cutter head with two edges is used.

A high-speed profiling table is used for finer work, the top of which measures 3 x 5 ft and has an air-driven cutter head operating at speeds up to 22,000 r.p.m. One-piece tool heads with a diameter of $\frac{1}{8}$ in. or even less, are used. This machine was designed and built in the Vallée plant.

The end cap used for profiling is made of wood and Masonite, or metal and Masonite. The Masonite being cut to exact shape and bearing against a live or dead collar around the base of the outer shell, depending on the set-up.

An example of this is found in the profiling of the battery compartment flap, which consists of two shaped extrusions. The side flanges of this part are released easily by changing the angle of the length of the extrusion. The flange being left half depth at certain angles, and three fourths depth at other angles. These gage plates of appropriate shape for attachment of the frame members that are later assembled to the run area.

This work is done with a steel or Monometal jig. The profile is adapted from a Gullinger and Larvignier wood shaper, with an auxiliary overhead arm to provide double bearing support for the tool spindle and to assure parallelism of the two edges, or flanges which are shaped simultaneously.

In feeding the work through the machine, the Minnabie guides at both top and bottom of the jig bear on top and bottom dead rollers around the cutting tool, supported from an independent bracket bolted to the table top. After the heavy cuts have been taken with this set-up the work is shifted to a second jig and another cut is taken, using a small cutting head of 3 in. diameter in cut narrow slots in portions of the beam flanges.

For light structural profiling work a 'live' collar is usually used around the outer head bearing directly on the Maxwode pattern edge of the profile block.

In meeting applications of products in the long shimmers—fifty gages placed to help fix the work rate lower in the walking edge with no surface sheet material. This gage is approximately 18 ft. long and of somewhat shape varying in width from about 6 to 12 in. to the material is not 9000 in. thick we had trouble with industry people. Most due to the edges of the material showing body under action of the cutting tool. We have solved the problem and at the same time greatly increased the speed of the work through developing a special handhold profile block sheet 18 ft. long, and using it in conjunction with an overhead system to

On the upper surface of the block we have mounted numerous locating pins which fit into rivet holes pre-drilled in the unclamped gunnet stock. In the bottom surface of this block are milled two slots extending the full length of the block. The slots exactly match the pattern of the two

In operation these slots fit into a pin in the top of the profiling table. The work is then drawn past the tool, dragging as it goes to conform with the bottom slots. By drawing it once past the tool it is now flat, and once in the other, the gasket plate is profiled accurately on both edges and is then ready for final assembly into the wing structure. In production it is possible to start with one slotted plate in an array to 15 or more and profile them all at once, which is a major benefit of laser cutting, and further assures regularity of the edges under action of the track.

[illegible]

**This is what we mean by
DEPENDABLE PERFORMANCE!**

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1. *Business Case*
 2. *Business Case*
 3. *Business Case*
 4. *Business Case*
 5. *Business Case*
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Abstract

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AVIATION
August, 1933
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Wright Researchers

(Continued from page 22)

from about the point X to change the angle of attack, in which case the angle indicated by the pointer P was added (subtracted when the angle was negative) to the angle of attack. We had a large scale at Dayton, not shown in the photograph, on the floor of the tunnel on which the instrument was mounted.

It will be observed that the surface S , being mounted on a pendulum, will maintain the same angle of attack at all positions of the arm AB , that the drag of the model, being balanced directly against the lift of the same surface, knowledge of the exact wind velocity is not required, that so long as the sides A , B are kept in a vertical position the weight of the surface S needs no counter-balancing; that the position of the center of pressure on the model S does not affect the readings, since the pointers always act on a radius equal to the length of the arm AB .

The force measuring arrangements may at first appear complicated, but to those acquainted with the modern techniques of wind tunnel research a number of outstanding features will immediately become apparent. The question of mounting the research instrument here involves obtaining a full-scale machine in the crucible of the whole situation, and in this respect the Wright brothers revealed not only superb understanding of mechanics but an astonishing appreciation of the pitfalls involved in model scale wind tunnel testing.

Oreilly Wright has given us a useful description of the whole process, which unfortunately I must reproduce in my own words. The lift, as noted before, was balanced directly against a set of resistance plates which had previously been calibrated against the drag of a square flat plate of known area at the same position as the air stream on the wings to be investigated. The drift was measured in per cent of the lift. Thus for the drag taken more easily to obtain deflections resulting from air-turbulence and non-uniform velocity in the air stream, although one had previously been taken to reduce those to a minimum. Still, however, no direct measurement of actual forces to be expected on the full-scale wings was available. This was accomplished by the use of a glider tested in 1907. This glider

had a wing scaled up from one of the models tested in the wind tunnel and by means of photo tests in winds of known velocity it was possible to determine the lift and drag as actual forces. These tests in effect calibrated one of the wings tested in the wind tunnel for full-scale conditions, and since the relation of the other wings to this wing was known from the wind tunnel measurements, the characteristics to be expected from all the wings when flown full size were directly indicated by the tested data.

They—Oreilly Wright, Glavin Wright, myself, I believe, one aspect of the Wright brothers' research work that has never been realized before—the fact that they adopted an experiment in the year 1905 that astronomically overestimated the Reynolds number to the Reynolds number encountered in flight. By the very description they reduced the critical Reynolds to one that the order of merit of the aerobically would not be seriously affected by Reynolds number, though the absolute values of the coefficients might change. It was not until about thirty years later that aerodynamic research indicated that question and showed that at least for a number of critical aerodynamic characteristics the assumption was valid.

One other point of interest in this connection deserves note. Using the average flight measurements on their 1902 glider and the wind tunnel data on models, the Wright brothers calculated that the total drag of a square flat plate of large area was equal to the square of the speed in miles per hour multiplied by the coefficient 0.0033. At the time, Oreilly Wright has told me, they had their choice

among figures varying from 0.0035 to 0.0038, based on the values of other investigators. Their subsequent research has led to general adoption of the figure 0.0033 for the coefficient in question, truly astounding agreement with the value reached by the Wright brothers through averaging a series of results obtained in flight measurements on a glider in the year 1902.

The results of the investigation on the aspect ratio (to wings) of various cases are shown in Fig. 5. These were reproduced directly from the original plate made in December of 1901. The first line of interest is the scales employed, which show the lift in per cent of the maximum of a square flat plate and the maximum lift coefficient of the wing in per cent of its lift. The shape of the curves is sufficiently familiar to all concerned with aerodynamics that they need no further classification. Models 7, 8 and 9, it will be noted, show the effects of center lift which we have later become familiar in more detail. It presents non-dimensional terms the drag coefficient of a square flat plate is approximately 14, so the figure shows that the maximum lift coefficient obtained with the smaller air aerobically varied from about 1.3 to about 1.5, values clearly in agreement with our modern knowledge of aerobically. With this information it is easily evident that the Wright brothers, through their aerodynamic research in 1901, proved a clear physical conception of the behavior of wings on which to base the design of their successful "flying machine." These subsequent flight progress in the field of aerobically and the technical and scientific achievements that have followed I need not recount.

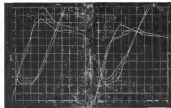


Fig. 5. Reproduction from original plate of data (1901)

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Simplifying Stress Analysis

(Continued from page 47)

The resultant of this flow is obviously equal to S and acts in the plane of the web.

Now if we replace the vertical web by a curved one, as in Fig. 3 (b), the shear flow is given by the same formula, but in this case the resultant S acts at a distance a to the left of the original position. Therefore we must apply the shear load S at this point if we do not want to twist the beam. The distance a is easily determined by the formula:

$$a = 2I(A)/S \quad (7)$$

which means that a is twice the average "height" of the enclosed area (A) between the shear web and the line joining the two flanges. The proof of this formula is very simple and can easily be shown by taking moments about any point.

The second important formula for shear flow is that commonly referred to as the "shear stress analogy," or by various other names. Perhaps it would be helpful to assign a more definitive name such as "torque-flow," which will be used in this paper. The torque-flow formula is:

$$q = \frac{T}{2A} \quad (8)$$

where T is the torque and A is the area of the section enclosed by the shell or ribs. This point is shown by Fig. 3 (c), representing a cross-section through a shell carrying the torque T . The resulting shear flow will be substantially constant if the ends of the box or tube are free to warp.

The third useful formula for shear flow is a more general form of the well-known formula for "horizontal" shear stress. If we assume that axial stresses due to bending agree with the classical beam theory ($\sigma = My/I$), the shear flow at any point on the shear web is given by the formula:

$$q = \frac{S}{I} \quad (9)$$

where S is the shear load, I is the "static moment" of the

area (as defined in any text book), I is the moment of inertia of the section.

Equation (4), incidentally, yields the simple equation (1) whenever the flange material is assumed to be concentrated so as to take all the axial loads due to bending.

Box Beam Analysis

A very helpful simplification in the stress analysis of single-cell beams was pointed out by R. S. Hirsch in his Letter to the Editor, in the July, 1958 issue of the *Journal of the Aeronautical Sciences*. It is often assumed that we need to find an "elastic axis" such that the application of shear loads along this axis will not twist the beam. While this is true for a multi-cell beam, it is entirely unnecessary for the single cell. The simplified procedure can be visualized by thinking of the beam as two separate structures, as shown in Fig. 4 (a) and (b). If the loads are not at any point it becomes possible to determine the shear flow due to direct shear by eq. 4, assuming that the loads thereby applied. This condition is illustrated by Fig. 4 (b). The location of the resultant shear, R , can be determined by taking moments of the shear flow loads about any convenient point. This can be done by dividing the periphery of the shell into small sections (which may be assumed to be straight, with little error), or by a graphical method such as suggested by Gilbert in the February, 1958, issue of the *Journal of the Aeronautical Sciences*. Now if we assume that the net shell has no torsional rigidity (which is practically true), it will be unable to carry a vertical shear load unless the loads are applied as so to coincide with the resultant of the shear flow (indicated by the distance a in Fig. 4 (b)). If the actual shear load S is applied at some other point, such as indicated by the distance b , it will be necessary to insert a torque equal to Sb . This torque can be resisted by the swept shell, as indicated in Fig. 4 (c), and the resulting shear flow is readily determined by the same-flow formula (eq. 9). To get the final result, it is only necessary to add, algebraically,

the shear flow due to direct shear (in the web shell) to the flow due to torque (in the swept shell).

As pointed out by Hirsch, this method of solution appears to solve a virtually indeterminate problem without recourse to load-carry or strain-energy methods. Actually, the only part of the problem that is statically indeterminate is the distribution of axial loads due to bending and this is taken care of by the usual beam formula. Even when the classical beam theory is modified by stress assumptions to cover the effects of "shear lag," the problem of shear flow remains statically determinate and can be solved by the foregoing methods, provided that the stresses due to bending are known or can be assumed.

A final simplification of single-cell analysis methods can be obtained by considering the load-carry method with the shear method of "cutting" the cell. Instead of determining the location of the resultant of the shear flow in the "cut" beam it is only necessary to obtain the moment of this flow about the load-axis. This torque may then be added to the external torque about the load-axis before applying the torque-flow formula. This process completely eliminates any necessity for the determination of an "elastic" axis or a shear axis. In stress analysis the computations can be carried out most conveniently by working with one load, such as 1000 lbs. for shear load. This permits much of the analysis work to be done independently of the determination of the loads for different design conditions.

Unfortunately there seems to be no immediate prospect of simplifying the multi-cell beam analysis to a statically determinate form. However, usage of the above principles can still be applied, as the problem is essentially the same except for some additional computations based on the principles of strain-energy or constant deformations. A careful study of the case in question will often show that more simplifying relationships exist, by which the problem can be reduced to statically-determinate form with little error. In view of the large number of conditions and assumptions that must be investigated for a modern large airplane structure, it is very important, even at a standpoint of cost and time, that analysis methods be kept as simple as possible. Refinements in stress analysis procedure should not get too far out of step with the other factors that influence the strength of the airplane.

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August, 1937
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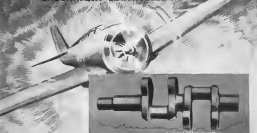
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49. Business Latin	50. Business Spanish	51. Business Portuguese	52. Business Russian
53. Business Chinese	54. Business Hindi	55. Business Urdu	56. Business Persian
57. Business Arabic	58. Business Hebrew	59. Business Greek	60. Business Latin
61. Business Spanish	62. Business Portuguese	63. Business Russian	64. Business Chinese
65. Business Hindi	66. Business Urdu	67. Business Persian	68. Business Arabic
69. Business Hebrew	70. Business Greek	71. Business Latin	72. Business Spanish
73. Business Portuguese	74. Business Russian	75. Business Chinese	76. Business Hindi
77. Business Urdu	78. Business Persian	79. Business Arabic	80. Business Hebrew
81. Business Greek	82. Business Latin	83. Business Spanish	84. Business Portuguese
85. Business Russian	86. Business Chinese	87. Business Hindi	88. Business Urdu
89. Business Persian	90. Business Arabic	91. Business Hebrew	92. Business Greek
93. Business Latin	94. Business Spanish	95. Business Portuguese	96. Business Russian
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Terms: 1. **By mail** 2. **By day school** 3. **By correspondence** 4. **By day school** 5. **By correspondence** 6. **By day school** 7. **By correspondence** 8. **By day school** 9. **By correspondence** 10. **By day school** 11. **By correspondence** 12. **By day school** 13. **By correspondence** 14. **By day school** 15. **By correspondence** 16. **By day school** 17. **By correspondence** 18. **By day school** 19. **By correspondence** 20. **By day school** 21. **By correspondence** 22. **By day school** 23. **By correspondence** 24. **By day school** 25. **By correspondence** 26. **By day school** 27. **By correspondence** 28. **By day school** 29. **By correspondence** 30. **By day school** 31. **By correspondence** 32. **By day school** 33. **By correspondence** 34. **By day school** 35. **By correspondence** 36. **By day school** 37. **By correspondence** 38. **By day school** 39. **By correspondence** 40. **By day school** 41. **By correspondence** 42. **By day school** 43. **By correspondence** 44. **By day school** 45. **By correspondence** 46. **By day school** 47. **By correspondence** 48. **By day school** 49. **By correspondence** 50. **By day school** 51. **By correspondence** 52. **By day school** 53. **By correspondence** 54. **By day school** 55. **By correspondence** 56. **By day school** 57. **By correspondence** 58. **By day school** 59. **By correspondence** 60. **By day school** 61. **By correspondence** 62. **By day school** 63. **By correspondence** 64. **By day school** 65. **By correspondence** 66. **By day school** 67. **By correspondence** 68. **By day school** 69. **By correspondence** 70. **By day school** 71. **By correspondence** 72. **By day school** 73. **By correspondence** 74. **By day school** 75. **By correspondence** 76. **By day school** 77. **By correspondence** 78. **By day school** 79. **By correspondence** 80. **By day school** 81. **By correspondence** 82. **By day school** 83. **By correspondence** 84. **By day school** 85. **By correspondence** 86. **By day school** 87. **By correspondence** 88. **By day school** 89. **By correspondence** 90. **By day school** 91. **By correspondence** 92. **By day school** 93. **By correspondence** 94. **By day school** 95. **By correspondence** 96. **By day school** 97. **By correspondence** 98. **By day school** 99. **By correspondence** 100. **By day school**



ABOUT WHAT *Holds* THINGS Together:

Do you know that the means of joining aluminum parts used in aircraft structures are being developed presently by intensive experimentation and research?



Rivets, which look so simple, are being improved. Investigation of the metal itself has resulted in a fine-grained material which flows more

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in production, but also because these methods help attain the very much desired smooth skin surface.

Now that there include the more important metal-joining techniques. This results in given the designer a wide range of possibilities in using alloys of Aluminum. And it facilitates maintenance, because repairs can be made by standard methods, rather than in all completely-enclosed shops.

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ALCOA ALUMINUM

AVIATION
August 1939
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Thompson Valves SERVE IN ANOTHER HISTORIC FLIGHT

The Yankee Clipper linked the old and new worlds on its initial flight in regular airline service May 21, 1938. The Yankee Clipper and its sister ship, the Atlantic Clipper now in trans-Atlantic mail and passenger service, are each powered by four 1800 H. P. Wright Twin Ram Cyclone engines, using 112 Thompson Valves.

SOME OTHER FAMOUS FLIGHTS WITH THOMPSON VALVES

Boeing Stearman's World's Closed Course Record—319.430 M. P. H. Stearman
Pittsburg Race 1938

Boeing Stearman—New York to Dallas—July 18, 1938

Howard Hughes Around the World Record—18,126 miles in 3 days, 15 hours, 51 minutes—July, 1931

Donnell and John Lusk—New York to Atlantic City—May 1937

Howard Hughes Lead Plane Speed Record—221 M. P. H.—November 12, 1935

Al and Fred Earl's Endurance Record—514 hours—16 minutes—June/July 1933

Stearman Stearman, New York to Chicago, Closed Course of 17 hours, 23 minutes—2400 miles—New York to Chicago—April 11, 1931

Dr. Charles Kettering-Smith, and Capt. P. G. Taylor—Baltimore to Columbia—November, 1934

World's Speed Record—489.90 M. P. H.—October, 1931—First Avialet at Lindbergh Field

20,000 mile World Tour—December 1933—Col. and Mrs. Chas. A. Lindbergh

Atlantic Route South Pole Flight—September, 1931—First South Pole Expedition 1929-1931, North Pole Expedition 1926

James Maitland—New York to Iceland, Shetland—June 1933

Avialet Enduro—First trans Atlantic flight by a woman—May 26, 1931

Kettering and Prichard—Japan to U. S. A.—New York Route—October 1934

Dr. A. Lusk, U. S. H. Atlantic Record by Lead Plane—10,800 feet—June 8, 1938

Lindbergh New York to Paris Spirit—May 20, 1927



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